



QUANSER
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Rotary Flexible Link

Workbook

FLEXGAGE

Instructor Version

Quanser Inc.

2011

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1 INTRODUCTION

The objective of this experiment is to control the position of the servo while minimizing the motions the flexible rotary link.

Topics Covered

- Modeling the Rotary Flexible Link using Lagrange.
- Find the linear state-space model of the system.
- Do some basic model validation.
- Design an state-feedback controller using Linear-Quadratic Regulator (LQR) algorithm through simulation.
- Implement the designed LQR controller on the device.
- Compare the simulated and measured closed-loop results.
- Assess the behaviour of implementing a partial-state feedback controller.

Prerequisites

In order to successfully carry out this laboratory, the user should be familiar with the following:

- Basics of [Simulink®](#).
- Transfer function fundamentals.
- State-space modeling, e.g., obtaining state equations from a set of differential equations.
- *SRV02 QUARC Integration Laboratory* ([2]) in order to be familiar using [QUARC®](#) with the SRV02.

Completion Time

The approximate times to complete each section is summarized in Table 1.

Section	Time (min)
Modeling Pre-lab (Section 2.2)	90 min
Modeling In-lab: Finding Stiffness (Section 2.3.1)	45 min
Modeling In-lab: Model Validation (Section 2.3.2)	75 min
Results (Section 2.4)	15 min
Control Pre-lab (Section 3.3)	45 min
Control In-lab: Simulation (Section 3.4.1)	60 min
Control In-lab: FSF Implementation (Section 3.4.2)	60 min
Control In-lab: PSF Implementation (Section 3.4.3)	30 min
Results (Section 3.5)	15 min

Table 1: Approximate Time to Complete

2 MODELING

2.1 Background

2.1.1 Model

The Rotary Flexible Link model is shown in Figure 2.1. The base of the flexible link is mounted on the load gear of the SRV02 system. The servo angle, θ , increases positively when it rotates counter-clockwise (CCW). The servo (and thus the link) turn in the CCW direction when the control voltage is positive, i.e., $V_m > 0$.

The flexible link has a total length of L_l , a mass of m_l , and its moment of inertia about the center of mass is J_l . See the *Rotary Flexible Link User Manual* (in [7]) for the values of these parameters. The deflection angle of the link is denoted as α and increases positively when rotated CCW.

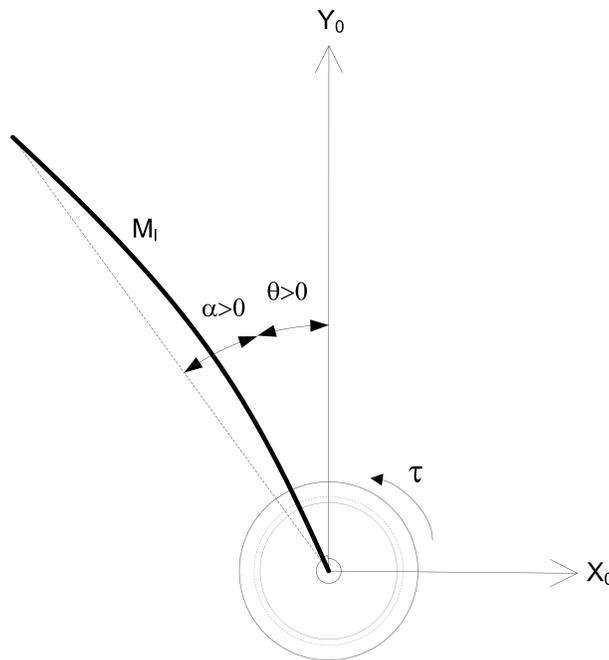


Figure 2.1: Rotary Flexible Link Angles

The flexible link system can be represented by the diagram shown in Figure 2.2. Our control variable is the input servo motor voltage, V_m . This generates a torque, τ , at the load gear of the servo that rotates the base of the link. The viscous friction coefficient of the servo is denoted by B_{eq} . This is the friction that opposes the torque being applied at the servo load gear. The friction acting on the link is represented by the viscous damping coefficient B_l . Finally, the flexible link is modeled as a linear spring with the stiffness K_s .

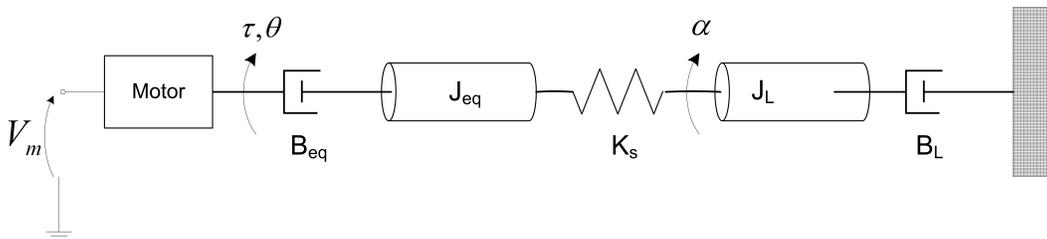


Figure 2.2: Rotary Flexible Link Model

2.1.2 Finding the Equations of Motion

Instead of using classical mechanics, the Lagrange method is used to find the equations of motion of the system. This systematic method is often used for more complicated systems such as robot manipulators with multiple joints.

More specifically, the equations that describe the motions of the servo and the link with respect to the servo motor voltage, i.e. the dynamics, will be obtained using the Euler-Lagrange equation:

$$\frac{\partial^2 L}{\partial t \partial \dot{q}_i} - \frac{\partial L}{\partial q_i} = Q_i \quad (2.1)$$

The variables q_i are called *generalized coordinates*. For this system let

$$q(t)^\top = [\theta(t) \ \alpha(t)] \quad (2.2)$$

where, as shown in Figure 2.2, $\theta(t)$ is the servo angle and $\alpha(t)$ is the flexible link angle. The corresponding velocities are

$$\dot{q}(t)^\top = \left[\frac{\partial \theta(t)}{\partial t} \quad \frac{\partial \alpha(t)}{\partial t} \right] \quad (2.3)$$

Note: The dot convention for the time derivative will be used throughout this document, i.e., $\dot{\theta} = \frac{d\theta}{dt}$ and $\dot{\alpha} = \frac{d\alpha}{dt}$. The time variable t will also be dropped from θ and α , i.e., $\theta := \theta(t)$ and $\alpha := \alpha(t)$.

With the generalized coordinates defined, the Euler-Lagrange equations for the rotary flexible link system are

$$\frac{\partial^2 L}{\partial t \partial \dot{\theta}} - \frac{\partial L}{\partial \theta} = Q_1 \quad (2.4)$$

and

$$\frac{\partial^2 L}{\partial t \partial \dot{\alpha}} - \frac{\partial L}{\partial \alpha} = Q_2 \quad (2.5)$$

The Lagrangian of a system is defined

$$L = T - V \quad (2.6)$$

where T is the total kinetic energy of the system and V is the total potential energy of the system. Thus the Lagrangian is the difference between a system's kinetic and potential energies.

The generalized forces Q_i are used to describe the non-conservative forces (e.g., friction) applied to a system with respect to the generalized coordinates. In this case, the generalized force acting on the rotary arm is

$$Q_1 = \tau - B_{eq} \dot{\theta} \quad (2.7)$$

and acting on the link is

$$Q_2 = -B_l \dot{\alpha}. \quad (2.8)$$

The torque applied at the base of the rotary arm (i.e., at the load gear) is generated by the servo motor as described by the equation

$$\tau = \frac{\eta_g K_g \eta_m k_t (V_m - K_g k_m \dot{\theta})}{R_m}. \quad (2.9)$$

See [4] for a description of the corresponding SRV02 parameters (e.g. such as the back-emf constant, k_m). The servo damping (i.e. friction), B_{eq} , opposes the applied torque. The flexible link is not actuated, the only force acting on the link is the damping, B_l .

Again, the Euler-Lagrange equations is a systematic method of finding the equations of motion (EOMs) of a system. Once the kinetic and potential energy are obtained and the Lagrangian is found, then the task is to compute various derivatives to get the EOMs.

2.1.3 Potential and Kinetic Energy

Kinetic Energy

Translational kinetic equation is defined as

$$T = \frac{1}{2}mv^2, \quad (2.10)$$

where m is the mass of the object and v is the linear velocity.

Rotational kinetic energy is described as

$$T = \frac{1}{2}J\omega^2 \quad (2.11)$$

where J is the moment of inertia of the object and ω is its angular rate.

Potential Energy

Potential energy comes in different forms. Typically in mechanical system we deal with *gravitational* and *elastic* potential energy. The *relative* gravitational potential energy of an object is

$$V_g = mg\Delta h, \quad (2.12)$$

where m is the object mass and Δh is the change in altitude of the object (from a reference point). The potential energy of an object that rises from the table surface (i.e., the reference) up to 0.25 meter is $\Delta h = 0.25 - 0 = 0.25$ and the energy stored is $V_g = 0.25mg$.

The equation for elastic potential energy, i.e., the energy stored in a spring, is

$$V_e = \frac{1}{2}K\Delta x^2 \quad (2.13)$$

where K is the spring stiffness and Δx is the linear or angular change in position. If an object that is connected to a spring moves from its initial reference position to 0.1 m, then the change in displacement is $\Delta x = 0.1 - 0 = 0.1$ and the energy stored equals $V_e = 0.005K$.

2.1.4 Linear State-Space Model

The linear state-space equations are

$$\dot{x} = Ax + Bu \quad (2.14)$$

and

$$y = Cx + Du \quad (2.15)$$

where x is the state, u is the control input, A , B , C , and D are state-space matrices. For the Rotary Flexible Link system, the state and output are defined

$$x^T = [\theta \ \alpha \ \dot{\theta} \ \dot{\alpha}] \quad (2.16)$$

and

$$y^T = [x_1 \ x_2]. \quad (2.17)$$

In the output equation, only the position of the servo and link angles are being measured. Based on this, the C and D matrices in the output equation are

$$C = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \end{bmatrix} \quad (2.18)$$

and

$$D = \begin{bmatrix} 0 \\ 0 \end{bmatrix}. \quad (2.19)$$

The velocities of the servo and link angles can be computed in the digital controller, e.g., by taking the derivative and filtering the result through a high-pass filter.

2.1.5 Free-Oscillation of a Second Order System

The free-oscillatory equation of motion of a second-order system described by

$$J\ddot{x} + B\dot{x} + Kx = 0 \quad (2.20)$$

is shown in Figure 2.3. Assuming the initial conditions $x(0^-) = x_0$ and $\dot{x}(0^-) = 0$, the Laplace transform of Equation 2.20 is

$$X(s) = \frac{\frac{x_0}{J}}{s^2 + \frac{B}{J}s + \frac{K}{J}} \quad (2.21)$$

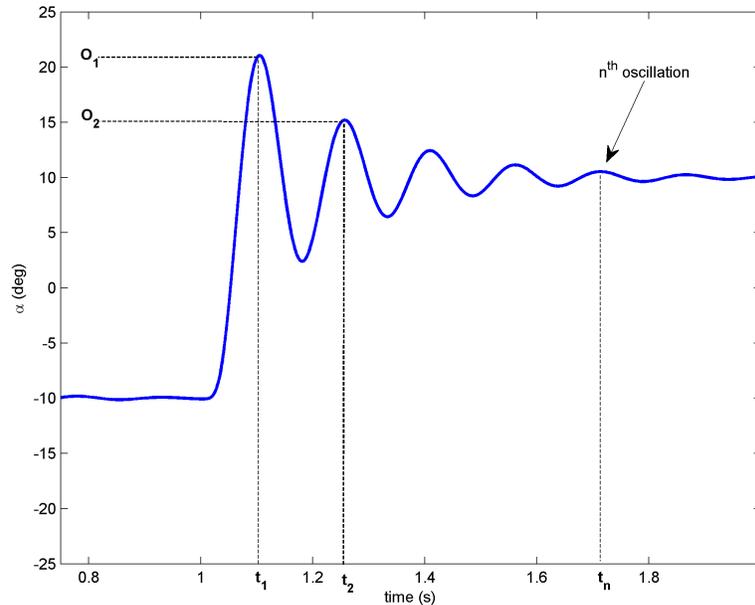


Figure 2.3: Free Oscillation Response

The prototype second-order equation is defined

$$s^2 + 2\zeta\omega_n s + \omega_n^2, \quad (2.22)$$

where ζ is the damping ratio and ω_n is the natural frequency. Equating the characteristic equation in 2.21 to this gives

$$\omega_n^2 = \frac{K}{J} \quad (2.23)$$

and

$$2\zeta\omega_n = \frac{B}{J} \quad (2.24)$$

Based on the measured damping ratio and natural frequency, the friction (or stiffness) of the system is

$$K = J\omega_n^2 \quad (2.25)$$

and the viscous damping is

$$B = 2\zeta\omega_n J. \quad (2.26)$$

Finding the Natural Frequency

The period of the oscillations in a system response can be found using the equation

$$T_{osc} = \frac{t_{n+1} - t_1}{n} \quad (2.27)$$

where t_n is the time of the n^{th} oscillation, t_1 is the time of the first peak, and n is the number of oscillations considered. From this, the damped natural frequency (in radians per second) is

$$\omega_d = \frac{2\pi}{T_{osc}} \quad (2.28)$$

and the undamped natural frequency is

$$\omega_n = \frac{\omega_d}{\sqrt{1 - \zeta^2}}. \quad (2.29)$$

Finding the Damping Ratio

The damping ratio of a second-order system can be found from its response. For a typical second-order under-damped system, the subsidence ratio (i.e., decrement ratio) is defined as

$$\delta = \frac{1}{n} \ln \frac{O_1}{O_n} \quad (2.30)$$

where O_1 is the peak of the first oscillation and O_n is the peak of the n^{th} oscillation. Note that $O_1 > O_n$, as this is a decaying response.

The damping ratio is defined

$$\zeta = \frac{1}{\sqrt{1 + \frac{2\pi^2}{\delta}}} \quad (2.31)$$

2.2 Pre-Lab Questions

1. **A-2** Energy is stored in the flexible link, i.e., the "spring", as it flexes by an angle of α (see Figure 2.1). Find the potential energy of the flexible link. Use the parameters shown in Figure 2.2.

Answer 2.1

Outcome **Solution**

A-2 Using Equation 2.13, the elastic energy stored in the spring equals

$$V = \frac{1}{2}K_s\alpha^2 \quad (\text{Ans.2.1})$$

This is the total potential energy that is store in the system.

□ □ □

2. **A-2** Find the total kinetic energy of the system contributed by the rotary servo, θ , and the deflection in the link, α . Use the parameters shown in Figure 2.2.

Answer 2.2

Outcome **Solution**

A-2 Using the Equation 2.11, the total kinetic energy from the SRV02 rotating and the deflection of the link is

$$T = \frac{1}{2}J_{eq}\dot{\theta}^2 + \frac{1}{2}J_l(\dot{\theta} + \dot{\alpha})^2. \quad (\text{Ans.2.2})$$

□ □ □

3. **A-2** Compute the Lagrangian of the system.

Answer 2.3

Outcome **Solution**

A-2 Using Equation 2.6, the total kinetic energy from the SRV02 rotating and the deflection of the link is

$$L = \frac{1}{2}J_{eq}\dot{\theta}^2 + \frac{1}{2}J_l(\dot{\theta} + \dot{\alpha})^2 - \frac{1}{2}K_s\alpha^2. \quad (\text{Ans.2.3})$$

□ □ □

4. **A-1, A-2** Find the first Euler-Lagrange equation given in 2.4. Keep the equations in terms of applied torque, τ (i.e., not in terms of DC motor voltage). Also make sure your equations follow the general form: $J\ddot{x} + B\dot{x} + Kx = u$.

Answer 2.4

Outcome Solution

A-1 Compute the derivatives required by Equation 2.4 and substitute the generalized force, Q_1 , given in Equation 2.7.

A-2 The derivatives are

$$\begin{aligned}\frac{\partial L}{\partial \dot{\theta}} &= J_{eq}\dot{\theta} + J_l(\dot{\theta} + \dot{\alpha}) \\ \frac{\partial^2 L}{\partial t \partial \dot{\theta}} &= J_{eq}\ddot{\theta} + J_l(\ddot{\theta} + \ddot{\alpha}) \\ \frac{\partial L}{\partial \theta} &= 0.\end{aligned}$$

Substituting the above answers and Equation 2.7 into Equation 2.4 gives the first equation of motion

$$(J_{eq} + J_l)\ddot{\theta} + J_l\ddot{\alpha} + B_{eq}\dot{\theta} = \tau \quad (\text{Ans.2.4})$$

□ □ □

5. **A-1, A-2** Find the second Euler-Lagrange Equation 2.5.

Answer 2.5

Outcome Solution

A-1 Compute the derivatives required by Equation 2.5 and substitute the generalized force, Q_2 , given in Equation 2.8.

A-2 The derivatives are

$$\begin{aligned}\frac{\partial L}{\partial \dot{\alpha}} &= J_l(\dot{\theta} + \dot{\alpha}) \\ \frac{\partial^2 L}{\partial t \partial \dot{\alpha}} &= J_l(\ddot{\theta} + \ddot{\alpha}) \\ \frac{\partial L}{\partial \alpha} &= K_s\alpha.\end{aligned}$$

Substituting the above derivative and Equation 2.8 into Equation 2.5 gives the second equation of motion

$$J_l\ddot{\theta} + J_l\ddot{\alpha} + B_l\dot{\theta} + K_s\alpha = 0 \quad (\text{Ans.2.5})$$

□ □ □

6. **A-1, A-2** Find the equations of motion: $\ddot{\theta} = f_1(\theta, \dot{\theta}, \alpha, \dot{\alpha}, \tau)$ and $\ddot{\alpha} = f_2(\theta, \dot{\theta}, \alpha, \dot{\alpha}, \tau)$. Assume the viscous damping of the link is negligible, i.e., $B_l = 0$.

Answer 2.6

Outcome Solution

- A-1 Solve for $\ddot{\theta}$ and $\ddot{\alpha}$ in Ans.2.4 and Ans.2.5.
A-2 Subtract equation Ans.2.4 from Ans.2.5 and set $B_l = 0$ to obtain the first equation of motion

$$\ddot{\theta} = -\frac{B_{eq}}{J_{eq}}\ddot{\theta} + \frac{K_s}{J_{eq}}\alpha + \frac{1}{J_{eq}}\tau. \quad (\text{Ans.2.6})$$

Substitute this into Equation Ans.2.5

$$\begin{aligned} J_l \left(-\frac{B_{eq}}{J_{eq}}\ddot{\theta} + \frac{K_s}{J_{eq}}\alpha + \frac{1}{J_{eq}}\tau \right) + J_l\ddot{\alpha} + K_s\alpha &= 0 \\ -\frac{J_l B_{eq}}{J_{eq}}\ddot{\theta} + J_l\ddot{\alpha} + K_s \left(\frac{J_l}{J_{eq}} + 1 \right) \alpha &= -\frac{J_l}{J_{eq}}\tau \end{aligned}$$

Solve for $\ddot{\alpha}$ to obtain the second equation of motion

$$\ddot{\alpha} = \frac{B_{eq}}{J_{eq}}\dot{\theta} - K_s \left(\frac{J_l + J_{eq}}{J_l J_{eq}} \right) \alpha - \frac{1}{J_{eq}}\tau \quad (\text{Ans.2.7})$$

□ □ □

7. **A-1, A-2** Given state x defined in Equation 2.16, find the linear state-space matrices A and B .

Answer 2.7

Outcome Solution

- A-1 Substitute state x , defined in Equation 2.16, into the equations of motion, Ans.2.6 and Ans.2.7. Then find matrices A and B defined in 2.14.
A-2 From the defined state in Equation 2.16, it is given that $\dot{x}_1 = x_3$ and $\dot{x}_2 = x_4$. Substituting x into Equation Ans.2.6 and Equation Ans.2.7 gives

$$\dot{x}_3 = -\frac{B_{eq}}{J_{eq}}x_3 + \frac{K_s}{J_{eq}}x_2 + \frac{1}{J_{eq}}u. \quad (\text{Ans.2.8})$$

and

$$\dot{x}_4 = \frac{B_{eq}}{J_{eq}}x_3 - K_s \left(\frac{J_l + J_{eq}}{J_l J_{eq}} \right) x_2 - \frac{1}{J_{eq}}u. \quad (\text{Ans.2.9})$$

The A and B matrices in the $\dot{x} = Ax + Bu$ equation are

$$A = \begin{bmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ 0 & \frac{K_s}{J_{eq}} & -\frac{B_{eq}}{J_{eq}} & 0 \\ 0 & -\frac{K_s(J_l + J_{eq})}{J_l J_{eq}} & \frac{B_{eq}}{J_{eq}} & 0 \end{bmatrix} \quad (\text{Ans.2.10})$$

and

$$B = \begin{bmatrix} 0 \\ 0 \\ \frac{1}{J_{eq}} \\ -\frac{1}{J_{eq}} \end{bmatrix}. \quad (\text{Ans.2.11})$$

□ □ □

8. **A-1, A-2** Find the the natural frequency of the response shown in Figure 2.3 if the peak values for the first and fifth oscillation are: $t_1 = 1.12$ and $t_5 = 1.71$ seconds. Because the damping is low, assume the damped and undamped natural frequency are equivalent.

Answer 2.8

Outcome Solution

A-1 Since ζ is small, assume that $\omega_n = \omega_d$ and find the natural frequency using Equation 2.28

A-2 Using Equation 2.27, the period is

$$\begin{aligned} T_{osc} &= \frac{t_5 - t_1}{4} \\ &= \frac{1.71 - 1.12}{4} \\ &= 0.148 \end{aligned}$$

From Equation 2.28, the natural frequency is

$$\omega_n = \frac{2\pi}{0.0148} = 42.6 \text{ rad/s} \quad (\text{Ans.2.12})$$

or 6.76 Hz.

□□□

2.3 Lab Experiments

In the first part of this laboratory, the stiffness of the flexible link is determined by measuring its natural frequency. In the second part, the state-space model is finalized and validated against actual measurements.

2.3.1 Finding Stiffness

In Section 2.1.5 we found an equation describing the free-oscillation response of a second-order system. This can also be used to describe the response of the flexible link when initially perturbed and left to decay.

Physical Parameters for the Lab

In order to do some of the laboratory exercises, you will need these values:

$$\begin{aligned} B_{eq} &= 0.004 \text{ N} \cdot \text{m} / (\text{rad/s}) \\ J_{eq} &= 2.08 \times 10^{-3} \text{ kg} \cdot \text{m}^2 \\ m_l &= 0.065 \text{ kg} \\ L_l &= 0.419 \text{ m} \end{aligned}$$

Note: The equivalent viscous damping, B_{eq} , and moment of inertia, J_{eq} , parameters are for the SRV02 when there is no load (i.e., the parameter found in the SRV02 Modeling Laboratory was for servo with the disc load).

Experimental Setup

The `q_flexgage_id` Simulink diagram shown in Figure 2.4 is used to find the natural frequency of the flexible link. The QUARC blocks are used to interface with strain gage sensor of the system. For more information about QUARC, see Reference [2]. This model outputs the deflection angle of the link.

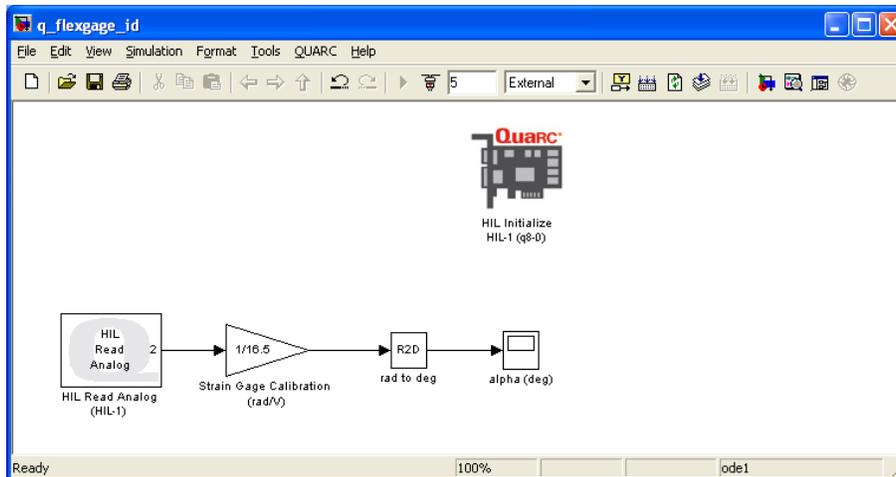


Figure 2.4: `q_flexgage_id` Simulink diagram used to find link stiffness

IMPORTANT: Before you can conduct this experiment, you need to make sure that the lab files are configured according to your system setup. If they have not been configured already, then you need to go to Section 4.2 to configure the lab files first.

1. In the `q_flexgage_id` Simulink diagram, go to QUARC | Build to build the QUARC controller.
2. Hold on the SRV02 load gear, go to QUARC | Start to run the controller, and then immediately perturb the flexible link. Keep holding the SRV02 base until the controller is done collecting the data. The *alpha (deg)* Scope should be reading a response similarly as shown in Figure 2.5. Note that the controller is set to run for 5 seconds.

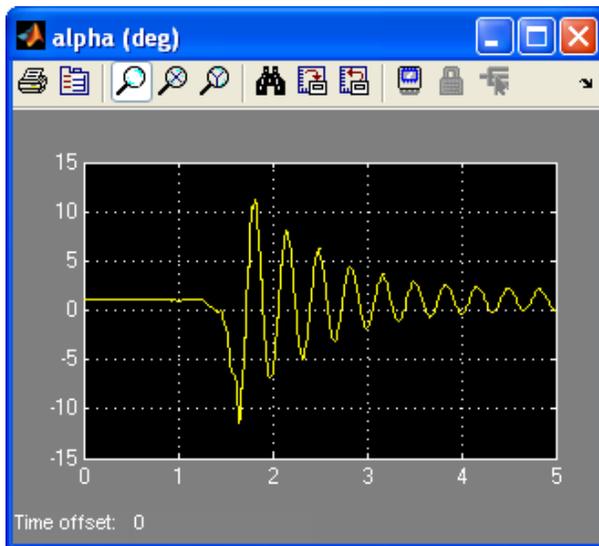


Figure 2.5: Typical Flexible Link Free-Oscillation Response

3. **B-5, K-2** After the controller stops (i.e., after 5 sec), the data is automatically saved in the Matlab workspace to the variable `data_alpha`. The time is stored in `data_alpha(:,1)` vector and the flexible link angle is stored the `data_alpha(:,2)` vector. Plot the response in a Matlab figure.

Answer 2.9

Outcome Solution

- B-5 If the experimental procedure was followed correctly, they should have generate a figure similar to Figure 2.6.
- K-2 Using the Matlab command `plot(data_alpha(:,1),data_alpha(:,2))` will generate a figure similarly as shown in Figure 2.6.

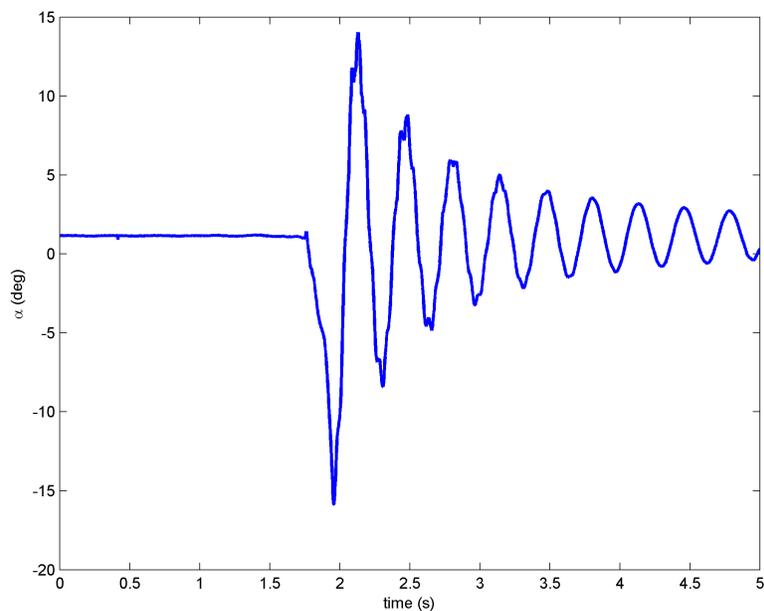


Figure 2.6: Typical Flexible Link Free Oscillation Response

□ □ □

4. **B-7** Find the natural frequency of the link. As in Pre-Lab Question 8, the damping is low. Therefore assume the damped natural frequency (which is being measured) is equivalent to the undamped natural frequency.

Answer 2.10

Outcome **Solution**
 B-7 Using Equation 2.27, the period is

$$T_{osc} = \frac{3.8 - 2.1}{5} = 0.34 \text{ s}$$

From Equation 2.28, the natural frequency is

$$\omega_n = \frac{2\pi}{0.34} = 18.5 \text{ rad/s} \quad (\text{Ans.2.13})$$

or 2.94 Hz.

□ □ □

5. **B-7** Calculate the stiffness of the flexible link, K_s .
Hint: Assume the link is a rod when finding the moment of inertia of the link, J_l .

Answer 2.11

Outcome **Solution**
 B-7 Recall the stiffness equation defined in 2.25 where the inertia and natural frequency are required. The moment of inertia of a rod pivoting about its end is defined: $J = ml^2/3$. Using the link mass, m_l , and length, L_l parameters given above, the moment of inertia of the link is

$$J_l = \frac{m_l L_l^2}{3} = 0.0038 \text{ kg} \cdot \text{m}^2. \quad (\text{Ans.2.14})$$

To get the measured stiffness, substitute the calculated inertia and the measured frequency, which was found in Ans.2.13, into Equation 2.25

$$K_s = 0.0038(18.5)^2 = 1.3 \text{ N m/rad}. \quad (\text{Ans.2.15})$$

□ □ □

2.3.2 Model Validation

By running a simulation and the actual device in parallel, we can verify whether the dynamic model (which drives the simulation) accurately represents our system.

Experimental Setup

The q_flexgag_val Simulink diagram shown in Figure 2.7 applies either a step or pulse input to both the Quanser Flexible Link hardware and to the Flexible Link model and reads the servo and link angles. The *SRV02 Flexible Link* subsystem contains the QUARC blocks that interface to the actual hardware. The Simulink *State-Space* block reads the *A*, *B*, *C*, and *D* state-space matrices that are loaded in the Matlab workspace. This model outputs the deflection angle of the link.

IMPORTANT: Before you can conduct this experiment, you need to make sure that the lab files are configured according to your system setup. If they have not been configured already, then go to Section 4.3 to configure the lab files first.

1. Run the *setup_flexgag.m*.

Answer 2.12

Outcome Solution

K-3 In the script, enter the A and B matrices in Ans.2.10 and Ans.2.11 and the C and D matrices in 2.18 and 2.19 that were found in Section 2.2:

```
A = [0 0 1 0;  
     0 0 0 1;  
     0 Ks/Jeq -Beq/Jeq 0;  
     0 -Ks*(Jeq+Jl)/Jeq/Jl Beq/Jeq 0];
```

```
B = [0 0 1/Jeq -1/Jeq]';
```

```
C = eye(2,4);
```

```
D = zeros(4,2);
```

The solution is given in `FLEXGAGE_ABCD_eqns.m`.

□ □ □

6. **B-5** Run the `FLEXGAGE_ABCD_eqns_student.m` script to load the state-space matrices in the Matlab workspace. Show the numerical matrices that are displayed in the Matlab prompt.

Answer 2.13

Outcome Solution

B-5 If the model was developed and entered properly, they should appear as:

A =

```
      0      0      1.0000      0  
      0      0      0      1.0000  
      0 623.7468 -7.1971      0  
      0 -965.3452  7.1971      0
```

B =

```
      0  
      0  
 479.8052  
-479.8052
```

C =

```
      1      0      0      0  
      0      1      0      0
```

D =

```
      0  
      0
```

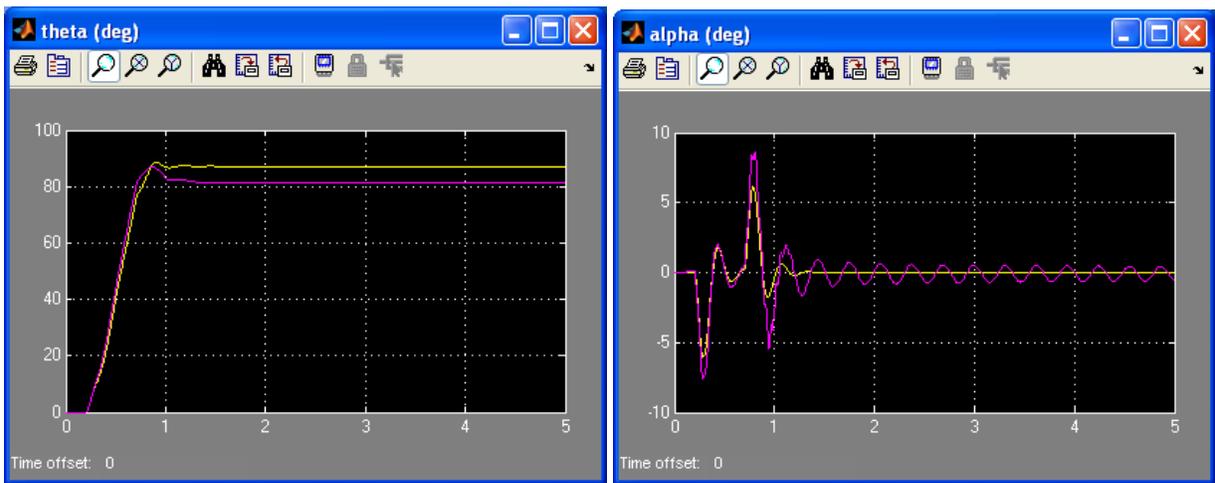
□ □ □

7. The input of the state-space model you found in Section 2.2 is the torque acting at the servo load gear (or the pivot of the flexible link). However, *we do not control torque directly - we control the servo input voltage*.

Recall the voltage-torque relationship given in Equation 2.9 in Section 2.1.2. In the *System Model* section of the `setup_flexgage.m` script, the actuator dynamics are added to your state-space matrices with the code:

```
Ao = A;
Bo = B;
B = eta_g*Kg*eta_m*kt/Rm*Bo;
A(3,3) = Ao(3,3) - Bo(3)*eta_g*Kg^2*eta_m*kt*km/Rm;
A(4,3) = Ao(4,3) - Bo(4)*eta_g*Kg^2*eta_m*kt*km/Rm;
```

8. Run the `setup_flexgage.m` script again so your Flexible Link model is based on dc motor voltage.
9. In the `q.flexgage_val` Simulink diagram, go to QUARC | Build to build the QUARC controller.
10. Make sure the area around the Flexible Link experiment is clear. Set the *Manual Switch* to the downward position to feed a Step input.
11. Go to QUARC | Start to run the `q.flexgage_val` controller. Typical scope responses are shown in Figure 2.8a and Figure 2.8b. The *theta (deg)* scope displays the simulated servo angle in yellow and the measured angle in purple. Similarly, the *alpha (deg)* scope shows the simulated link angle in yellow and the measured angle in purple.



(a) Servo Angle

(b) Flexible Link Angle

Figure 2.8: Typical Flexible Link Model Validation Response

12. If your simulation and measured response match, go to the next step. If they do not, then there is an issue with your model. Here are some issues to investigate:
 - Ensure the state-space model was entered properly in the script.
 - The stiffness, K_s , found in Section 2.3.1 is not correct. Review your calculations or redo the experiment.
 - Review your model derivation in Section 2.2, e.g., might be a mistake in solving the EOMs.
13. **B-5, K-2** The *theta (deg)* and *alpha (deg)* scopes save their data to the variables `data_theta` and `data_alpha`. For `data_theta`, the time is in `data_theta(:,1)`, the simulated servo angle is in `data_theta(:,2)`, and the measured SRV02 angle is in `data_theta(:,3)`. Similarly for the flexible link response saved in `data_alpha`. Plot the response in a Matlab figure.

Answer 2.14

Outcome Solution

- B-5 If the experimental procedure was followed correctly, they should have generated a figure similar to Figure 2.9.
- K-2 Use the Matlab *plot* command, you can generate a figure similarly as shown in Figure 2.9. Run the *plot_flexgage_val.m* script to plot this response.

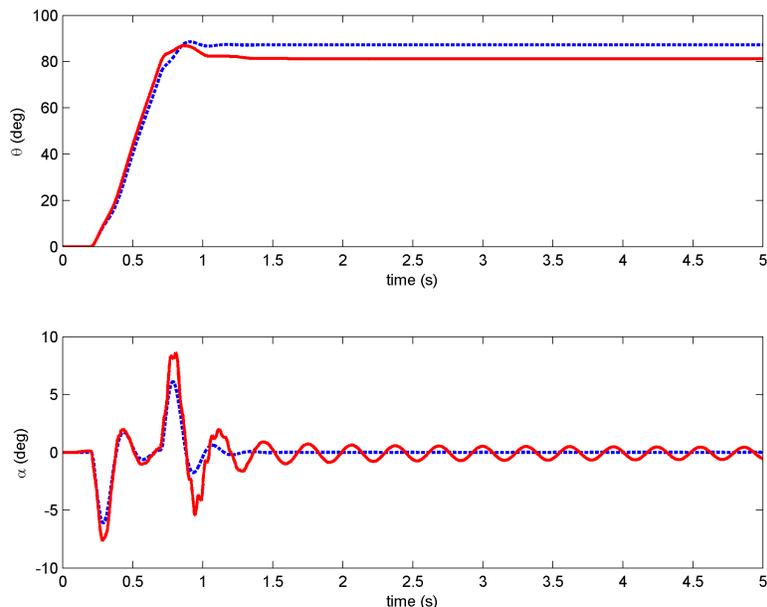


Figure 2.9: Typical Flexible Link Model Validation Response

□ □ □

14. **B-9** How well does your model represent the actual system? We want a model that is fairly representative but, having said that, keep in mind that no model is perfect. This is just a quick test to see how well your model represents the actual device. As shown in figures 2.8a and 2.8b, the simulation does not match the measured response perfectly.

Answer 2.15

Outcome Solution

- B-9 As shown in Figure 2.9, the model represents the system quite well. In the servo transient response (top plot), the model tracks the measured. But some discrepancy occurs in the steady-state (primarily due to the Coulomb friction and not having the correct viscous damping B_{eq}). The flexible link (bottom plot) matches well for the first few oscillation (when the amplitude is high) but doesn't capture the later part of the decaying response when the amplitude is low.

□ □ □

15. **B-8** Give at least one reason why the model does not represent the system accurately.

Answer 2.16

Outcome Solution

B-8

Here are some reasons:

- The model does not take Coulomb friction, i.e., stiction, into account.
- Flexible link is modeled as a linear spring. Actual link is nonlinear (e.g., has high-frequency models that are unmodeled).
- Servo viscous damping, B_{eq} , different for that servo.
- Neglecting the flexible link damping, B_l .

□ □ □

16. **K-1** In Matlab, find the open-loop poles (i.e., eigenvalues) of the system using the state-space matrix A that is loaded.

Note: These will be required for a pre-lab question in Section 3.3.

Answer 2.17

Outcome Solution

K-1

Using the Matlab command $\text{eig}(A)$, the open-loop poles of the system are: -24 , $-8.16 + j22.5$, $-8.16 - j22.5$, and 0 .

□ □ □

2.4 Results

B-6 Fill out Table 2 with your answers from your modeling lab results - both simulation and implementation.

Description	Symbol	Value	Unit
Finding Stiffness			
Natural frequency	ω_n	18.5	rad/s
Stiffness	K_s	1.3	N m/rad
Model Validation			
State-Space Matrix	A	$\begin{bmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ 0 & 623.7 & -7.20 & 0 \\ 0 & -965.3 & 7.20 & 0 \end{bmatrix}$	
State-Space Matrix	B	$\begin{bmatrix} 0 \\ 0 \\ 479.8052 \\ -479.8052 \end{bmatrix}$	
State-Space Matrix	C	$\begin{bmatrix} 0 & 1 & 0 & 0 \\ 1 & 0 & 0 & 0 \end{bmatrix}$	
State-Space Matrix	D	$\begin{bmatrix} 0 \\ 0 \end{bmatrix}$	
Open-loop poles	OL	{0,-24, -8.16+j22.5, -8.16-j22.5}	

Table 2: Results

3 CONTROL DESIGN

3.1 Specifications

The time-domain requirements are:

Specification 1: Servo angle settling time: $t_s \leq 0.5$ s.

Specification 2: Servo angle percentage overshoot: $PO \leq 7.5$ %.

Specification 3: Maximum link angle deflection: $|\alpha| \leq 10$ deg.

Specification 4: Maximum control effort / voltage: $|V_m| \leq 10$ V.

These specifications are to be satisfied when the rotary arm is tracking a ± 30 degree angle square wave.

3.2 Background

In Section 2.2, we found a linear state-state space model that represents the Rotary Flexible Link system. This model is used to investigate the stability properties of the Flexible Link system in Section 3.2.1. In Section 3.2.2, the notion of controllability is introduced. Using the Linear Quadratic Regular algorithm, or LQR, is a common way to find the control gain and is discussed in Section 3.2.3. Lastly, Section 3.2.4 describes the state-feedback control used to control the servo position while minimizing link deflection.

3.2.1 Stability

The stability of a system can be determined from its poles ([8]):

- Stable systems have poles only in the left-hand plane.
- Unstable systems have at least one pole in the right-hand plane and/or poles of multiplicity greater than 1 on the imaginary axis.
- Marginally stable systems have one pole on the imaginary axis and the other poles in the left-hand plane.

The poles are the roots of the system's characteristic equation. From the state-space, the characteristic equation of the system can be found using

$$\det(sI - A) = 0 \quad (3.1)$$

where $\det()$ is the determinant function, s is the Laplace operator, and I the identity matrix. These are the *eigenvalues* of the state-space matrix A .

3.2.2 Controllability

If the control input, u , of a system can take each state variable, x_i where $i = 1 \dots n$, from an initial state to a final state then the system is controllable, otherwise it is uncontrollable ([8]).

Rank Test The system is controllable if the rank of its controllability matrix

$$T = [B \ AB \ A^2B \ \dots \ A^nB] \quad (3.2)$$

equals the number of states in the system,

$$\text{rank}(T) = n. \quad (3.3)$$

3.2.3 Linear Quadratic Regular (LQR)

If (A,B) are controllable, then the Linear Quadratic Regular optimization method can be used to find a feedback control gain. Given the plant model in Equation 2.14, find a control input u that minimizes the cost function

$$J = \int_0^{\infty} x(t)'Qx(t) + u(t)'Ru(t) dt, \quad (3.4)$$

where Q and R are the weighting matrices. The weighting matrices affect how LQR minimizes the function and are, essentially, tuning variables.

Given the control law $u = -Kx$, the state-space in Equation 2.14 becomes

$$\begin{aligned} \dot{x} &= Ax + B(-Kx) \\ &= (A - BK)x \end{aligned}$$

3.2.4 Feedback Control

The feedback control loop that in Figure 3.1 is designed to stabilize the servo to a desired position, θ_d , while minimizing the deflection of the flexible link.

The reference state is defined

$$x_d = [\theta_d \ 0 \ 0 \ 0] \quad (3.5)$$

and the controller is

$$u = K(x_d - x). \quad (3.6)$$

Note that if $x_d = 0$ then $u = -Kx$, which is the control used in the LQR algorithm.

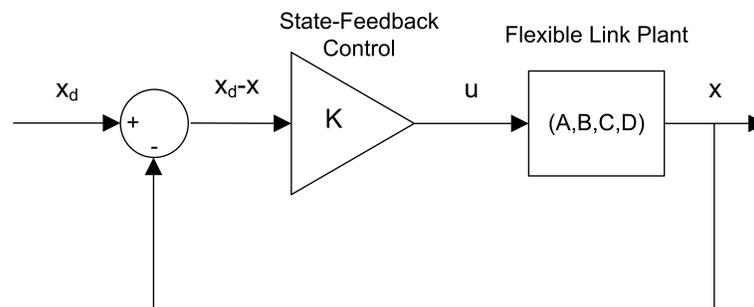


Figure 3.1: State-feedback control loop

3.3 Pre-Lab Questions

1. **A-1, A-3** Based on your analysis of the system in the Modeling Laboratory (Section 2.3), is the system stable, marginally stable, or unstable? From your experience in Section 2.3, does the stability you determined analytically match how the actual system behaves?

Answer 3.1

Outcome Solution

A-1 Recall from Step 16 in Section 2.3 the open-loop poles are -24, -8.16+j22.5, -8.16-j22.5, and 0. Because one pole is on the imaginary axis (and all the others are in the left-hand plane), the system is *marginally stable*. This is based on the definitions outlined in Section 3.2.1.

A-3 From the Modeling laboratory, rotating the servo back and forth does not cause the system to go unstable but it does make the link vibrate and introduces oscillations in the response. Due to these oscillations, the system cannot be categorized as stable but rather is marginally stable.

□ □ □

2. **A-1, A-2** Designing a controller with the Linear Quadratic Regular (LQR) technique is an iterative process. In software, you have to tune the Q and R matrices, generate the gain K using LQR, and either simulate the system or implement the control to see if you have the desired response. The relationship between changing Q and R and the closed-loop response is not evident. However, we can have a better idea on how changing the different elements in Q and R will effect the response. We will only be changing the diagonal elements in Q , thus let

$$Q = \begin{bmatrix} q_1 & 0 & 0 & 0 \\ 0 & q_2 & 0 & 0 \\ 0 & 0 & q_3 & 0 \\ 0 & 0 & 0 & q_4 \end{bmatrix}. \quad (3.7)$$

Since we are dealing with a single-input system, R is a scalar value. Using the Q and R defined, expand the cost function given in Equation 3.4.

Answer 3.2

Outcome Solution

A-1 Substitute Q given in Equation 3.7 above into Equation 3.4 and expand the equation.

A-2 The cost function becomes

$$\begin{aligned} J &= \int_0^{\infty} x' Q x + R u^2 dt \\ &= \int_0^{\infty} [x_1 \ x_2 \ x_3 \ x_4] \begin{bmatrix} q_1 & 0 & 0 & 0 \\ 0 & q_2 & 0 & 0 \\ 0 & 0 & q_3 & 0 \\ 0 & 0 & 0 & q_4 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ x_4 \end{bmatrix} + R u^2 dt \\ &= \int_0^{\infty} q_1 x_1^2 + q_2 x_2^2 + q_3 x_3^2 + q_4 x_4^2 + R u^2 dt \quad (\text{Ans.3.1}) \end{aligned}$$

□ □ □

3. **A-3** For the feedback control $u = -Kx$, the Linear-Quadratic Regular algorithm finds a gain K that minimizes the cost function J . Matrix Q sets the weight on the states and determines how u will minimize J (and hence how it generates gain K). From your solution in Question 2, explain how increasing the diagonal elements, q_i , effects the generated gain $K = [k_1 \ k_2 \ k_3 \ k_4]$.

Answer 3.3

Outcome Solution

A-3 Looking at Equation Ans.3.1, increasing q_i causes control input u to work harder to minimize state x_i , which will predominately increase k_i . If q_1 is increased then k_1 will increase to compensate for the larger weight placed on state x_1 . Depending on the model, changing a single q_i can effect multiple k_i gains because each state is not independent.

□ □ □

4. **A-3** Explain the effect of increasing R has on the generated gain, K .

Answer 3.4

Outcome Solution

A-3 Looking at Equation Ans.3.1, if R is increased then control input u has to work less to minimize J . In that case, LQR would generate a lower overall gain of K .

□ □ □

3.4 Lab Experiments

The control gain is designed using LQR through simulation first. Once a gain that satisfies the requirements is found, it is implemented on the actual Quanser Flexible Link system.

3.4.1 Control Simulation

Using the linear state-space model of the system and the designed control gain, the closed-loop response can be simulated. This way, we can test the controller and see if it satisfies the given specifications before running it on the hardware platform.

Experiment Setup

The *s.flexgag* Simulink diagram shown in Figure 3.2 is used to simulate the closed-loop response of the Flexible Link using the LQR control developed in Section 3.3.

The *Smooth Signal Generator* block generates a 0.33 Hz square wave (with amplitude of 1) that is passed through a *Rate Limiter* block to smooth the signal. The *Amplitude (deg)* gain block is used to change the desired servo position command. The state-feedback gain K is set in the *LQR Control* gain block and is read from the Matlab workspace. The Simulink *State-Space* block reads the A , B , C , and D state-space matrices that are loaded in the Matlab workspace.

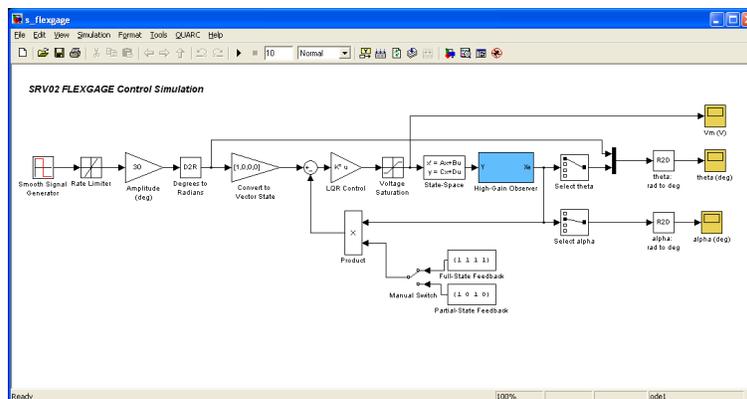


Figure 3.2: *s.flexgag* Simulink diagram used to simulate the state-feedback control

IMPORTANT: Before you can conduct this experiment, you need to make sure that the lab files are configured according to your system setup. If they have not been configured already, go to Section Section 4.4 to configure the lab files first. **Make sure the model you found in Section 2.3.2 is entered in FLEXGAGE_ABCD_eqns_student.m.**

1. Open *setup.flexgag.m* and go down to the *LQR Control* section shown here:

```
%% LQR Control
if strcmp ( CONTROL_TYPE , 'MANUAL' )
    % Set Q and R matrices to get desired response.
    Q = diag([1 1 1 1]);
    R = 1;
    [K,S,E] = lqr(A,B,Q,R);
```

The Q and R are initially set to the default values of:

$$Q = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix} \text{ and}$$

$$R = 1.$$

- These will not give you the desired response, but run the script to generate the default gain K . Enter the stiffness you found in Section 2.3.1 when prompted.
- Run `s_flexgag` to simulated the closed-loop response with this gain. See figures 3.3a, 3.3b, and 3.3c for the typical response.

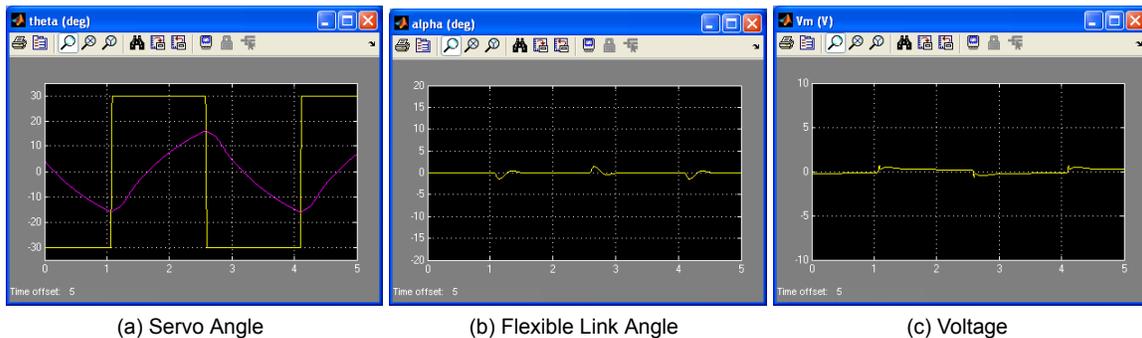


Figure 3.3: Default Simulated Closed-Loop Response

- B-2** If $Q = \text{diag}[q_1, q_2, q_3, q_4]$, vary each q_i independently and examine its effect on the gain and the closed-loop response. For example, when increasing q_3 , what happens to θ and α ? Vary each q_i by the same order of magnitude and compare how the new gain K changes compared to the original gain. Keep $R = 1$ throughout your testing. Summarize your results.

Note: Recall your analysis in pre-lab Question 3 where the effect of adjusting Q on the generated K was assessed generally by inspecting the cost function equation. You may find some discrepancies in this exercise and the pre-lab questions.

Answer 3.5

Outcome Solution

B-2 Let $K = [k_1 \ k_2 \ k_3 \ k_4]$. The elements effect the gains as follows:

- Increasing q_1 increases servo proportional gain k_1 . Makes response faster, i.e., decreases peak and settling time.
- q_2 does not have much effect.
- Increasing q_3 increases servo derivative gain k_3 . It also makes k_4 more positive. Minimizes overshoot of servo response but also slows it down.
- Increasing q_4 decreases the link proportional gain k_2 and derivative gain k_4 . Minimizes deflection of flexible link without affecting the servo related gains, k_1 and k_3 , significantly.

From this exercise, students should identify that the q_1 and q_4 elements have the greatest effect on the response.

□ □ □

- K-1** Find a Q and R that will satisfy the specifications given in Section 3.1. When doing this, don't forget to keep the dc motor voltage within ± 10 V. This control will later be implemented on actual hardware. Therefore,

make sure the actuator is not being saturated. Enter the weighting matrices, Q and R , used and the resulting gain, K .

Answer 3.6

Outcome Solution

K-1 One set of weighting matrices that yields adequate results are

$$Q = \begin{bmatrix} 150 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 3 \end{bmatrix} \quad (\text{Ans.3.2})$$

and

$$R = 1 \quad (\text{Ans.3.3})$$

which generates the gain

$$K = [12.2 \quad -23.0 \quad 1.39 \quad -0.311]. \quad (\text{Ans.3.4})$$

□□□

6. **B-5, K-3** Plot the responses from the θ (deg), α (deg), and V_m (V) scopes in a Matlab figure. When the QUARC controller is stopped, these scopes automatically save the last 5 seconds of their response data to the variables `data_theta`, `data_alpha`, and `data_vm`. For `data_theta`, the time is in `data_theta(:,1)`, the setpoint (i.e., desired SRV02 angle) is in `data_theta(:,2)`, and the simulated SRV02 angle is in `data_theta(:,3)`. In the `data_alpha` and `data_vm` variables, the `(:,1)` holds the time vector and the `(:,2)` holds the actual measured data.

Answer 3.7

Outcome Solution

B-5 If the experimental procedure was followed correctly, they should have generated a figure similar to Figure 3.4.

K-3 Use the Matlab `plot` command, you can generate a figure similarly as shown in Figure 3.4. Run the `meas_flexgag_spec.m` script after running the `s_flexgag.mdl` to plot this response.

□□□

7. **K-1, B-9** Measure the settling time and percent overshoot of the simulated servo response and the maximum link deflection. Does the response satisfy the specifications given in Section 3.1?

Answer 3.8

Outcome Solution

K-1 The settling time and overshoot measured from the top plot in Figure 3.4 are

$$t_s = 0.42 \text{ s}$$

and

$$PO = 2.8$$

The maximum deflection of the link is

$$|\alpha|_{\max} = 9.5 \text{ deg}$$

B-9 The specifications in Section 3.1 are satisfied while the input voltage is kept between ± 10 V. Use the `meas_flexgag_spec.m` script to find the specifications automatically after running `s_flexgag`.

□□□

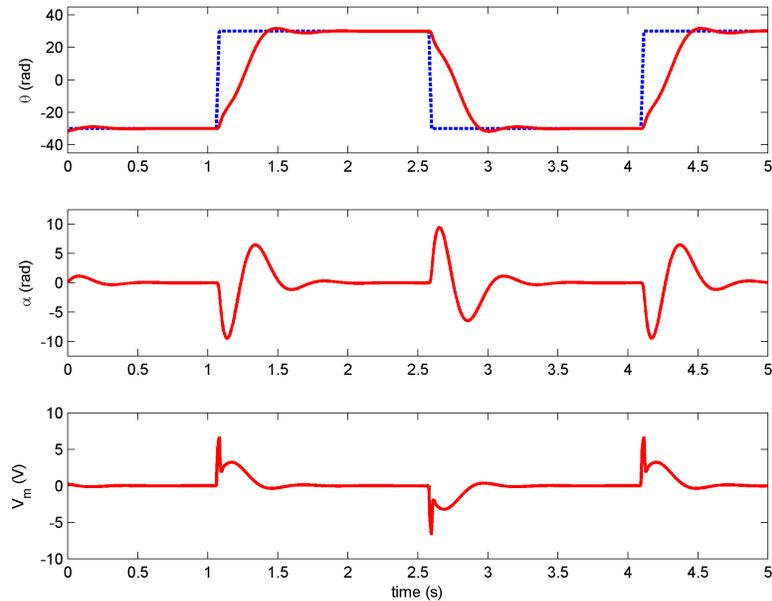


Figure 3.4: Simulated Closed-Loop Response

8. **B-4** Briefly explain the procedure you used to find Q and R .

Answer 3.9

Outcome Solution

B-4 Increased q_1 term until the settling time is satisfied. Then increased q_4 to minimize the deflection in the link angle.

□ □ □

3.4.2 Control Implementation

In this experiment, the servo position is controlled while minimizing the link deflection using the LQR-based control found in Section 3.4.1. Measurements will then be taken to ensure that the specifications are satisfied.

Experiment Setup

The $q_flexgag$ Simulink diagram shown in Figure 3.5 is used to run the state-feedback control on the Quanser Flexible Link system. The *SRV02 Flexible Link* subsystem contains QUARC blocks that interface with the DC motor and sensors of the system. The feedback developed in Section 3.4.1 is implemented using a Simulink *Gain* block.

IMPORTANT: Before you can conduct this experiment, you need to make sure that the lab files are configured according to your system setup. If they have not been configured already, then go to Section 4.5 to configure the lab files first.

1. Run the *setup_flexgag.m*.
2. Ensure the LQR controller you found in Section 3.4.1 is loaded, i.e., gain K .
3. Open the $q_flexgag$ Simulink diagram. Make sure the *Manual Switch* is set to the *Full-State Feedback* (upward) position.
4. Go to QUARC | Build to build the controller.

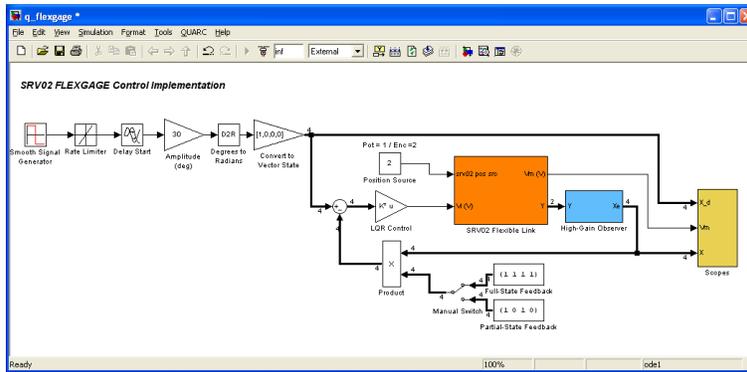


Figure 3.5: q.flexgage Simulink diagram used the model

5. Go to QUARC | Start to run the controller. The flexible link should be tracking the default ± 30 degree signal.
6. Stop the controller once you have obtained a representative response.
7. **B-5, K-3** Plot the responses from the θ (deg), α (deg), and V_m (V) scopes in a Matlab figure. Similarly as described in Section 3.4.1, the response data is saved in variables $data_theta$, $data_alpha$, and $data_vm$.

Answer 3.10

Outcome Solution

- B-5** If the experimental procedure was followed correctly, they should have generated a figure similar to Figure 3.6.
- K-3** Use the Matlab *plot* command, you can generate a figure similarly as shown in Figure 3.6. Run the *meas.flexgage.specs.m* script after running the *q.flexgage.mdl* with the LQR gain found in Equation Ans.3.4 to plot this response. Alternatively, this plot can be generated using the data stored in the *data.flexgage.fsf.theta.mat*, *data.flexgage.fsf.alpha.mat*, and *data.flexgage.fsf.vm.mat* files.

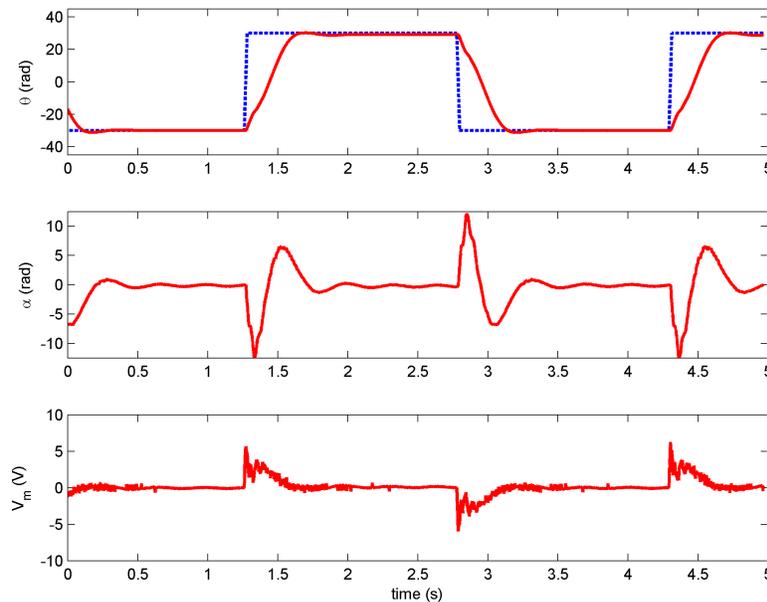


Figure 3.6: Flexible Link Closed-Loop Response

□ □ □

8. **K-1, B-9** Measure the settling time and percent overshoot of the measured servo response and the maximum link deflection. Does the response satisfy the specifications given in Section 3.1?

Answer 3.11

Outcome Solution

K-1 The specifications in Figure 3.6 are

$$\begin{aligned}t_s &= 0.43 \text{ s}, \\PO &= 1.8 \%, \\|\alpha|_{\max} &= 12.4 \text{ deg}, \\|V_m| &\leq 10 \text{ V}\end{aligned}$$

B-9 The specifications in Section 3.1 are not satisfied because the flexible link exceeds the 10 degree limit. Run `meas.flexgage.specs.m` script to find the specifications automatically (after running `q.flexgage` or from saved data).

□ □ □

9. **K-1** If the specifications have been met, then you can proceed to Section 3.4.3 or your report (ask your instructor). If they have not been met, then you need to tweak your controller. Using the principles you learnt in Section 3.4.1, tune the current Q and R matrices to satisfy the specifications. Enter the weighting matrices, Q and R , used and the resulting gain, K .

Answer 3.12

Outcome Solution

K-1 One set of weighting matrices that yields adequate results are

$$Q = \begin{bmatrix} 125 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 5 \end{bmatrix} \quad (\text{Ans.3.5})$$

and

$$R = 1 \quad (\text{Ans.3.6})$$

which generates the gain

$$K = [11.2 \quad -30.5 \quad 1.46 \quad -0.680]. \quad (\text{Ans.3.7})$$

□ □ □

10. **B-5, K-3** Plot the responses from the θ (deg), α (deg), and V_m (V) scopes in a Matlab figure of your newly *tuned* controller. Similarly as described in Section 3.4.1, the response data is saved in variables `data_theta`, `data_alpha`, and `data_vm`.

Answer 3.13

Outcome Solution

- B-5 If the student was able to tune the controller based on the exercises in Section 3.4.1, they should have generated a figure similar to Figure 3.7.
- K-3 Use the Matlab `plot` command, you can generate a plot like Figure 3.7. Run the `meas_flexgage_specs.m` script after running the `q_flexgage.mdl` with Equation Ans.3.7 to plot this response or using the data saved in `data_flexgage_fsf2_theta.mat`, `data_flexgage_fsf2_alpha.mat`, and `data_flexgage_fsf2_vm.mat` files.

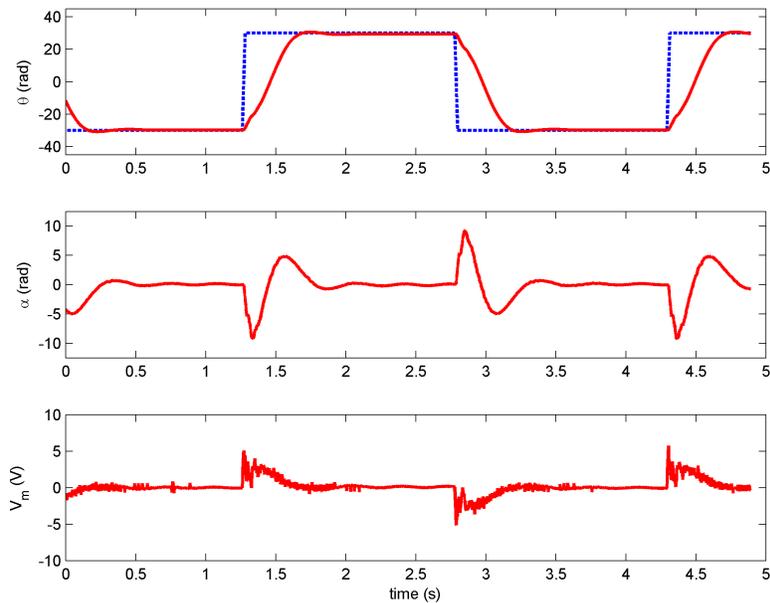


Figure 3.7: Flexible Link Closed-Loop Response - Tuned Controller

□ □ □

11. **K-1, B-9** Measure the settling time and percent overshoot of the simulated servo response and the maximum link deflection. Does the response satisfy the specifications given in Section 3.1?

Answer 3.14

Outcome Solution

- K-1 The specifications in Figure 3.6 are

$$\begin{aligned}t_s &= 0.48 \text{ s,} \\PO &= 1.9 \%, \\|\alpha|_{\max} &= 9.2 \text{ deg,} \\|V_m| &\leq 10 \text{ V}\end{aligned}$$

- B-9 The specifications in Section 3.1 are satisfied with the tuned controller. Run `meas_flexgage_spec.m` script to find the specifications automatically after running `q_flexgage` with gain K given in Equation Ans.3.7.

□ □ □

12. **B-4** Explain the procedure you used to tune your controller.

Answer 3.15

Outcome Solution

B-4 Increasing parameter q_4 from 3 to 5 dampens the link deflection but it also increases the speed of the servo response, i.e., resulting in a larger settling time. By decreasing q_1 from 150 down to 125, the servo response settling time is decreased within specification.

□ □ □

3.4.3 Implementing Partial-State Feedback Control

In this section, the partial-state feedback response of the system is assessed and compared with the full-state feedback control in Section 3.4.2.

1. Run the `setup_flexgage.m`.
2. Ensure the LQR control gain you settled on in Section 3.4.2 is loaded, i.e., gain K .
3. Open the `q.flexgage` Simulink diagram. Make sure the *Manual Switch* is set to the *Partial-State Feedback* (downward) position.
4. Go to QUARC | Start to run the controller. The flexible link should be tracking the default ± 30 degree signal.
5. Stop the controller once you have obtained a representative response.
6. **B-5, K-3** As in Section 3.4.2, attach a Matlab figure representing the SRV02 angle and flexible link angle response, as well as the input voltage.

Answer 3.16

Outcome Solution

B-5 If the experimental procedure was followed correctly, they should have generated a figure similar to Figure 3.8.

K-3 Use the Matlab `plot` command, you can generate a figure similarly as shown in Figure 3.8. Run the `meas_flexgage_specs.m` script after running the `q.flexgage.mdl` to plot this response using gain K defined in Equation Ans.3.7 and when the switch is on partial-state feedback. Alternatively, you can plot it using the data saved in `data_flexgage_psf_theta.mat`, `data_flexgage_psf_alpha.mat`, and `data_flexgage_psf_vm.mat` files.

□ □ □

7. **B-7** Examine the difference between the partial-state feedback (PSF) response and the full-state feedback (FSF) response. Explain why PSF control behaves this way by looking at the `q.flexgage` Simulink diagram.

Answer 3.17

Outcome Solution

B-7 By examining the feedback loop in `q.flexgage`, in PSF mode only the servo position and velocity states are fed back. Thus the full state in Equation 2.16 becomes $x = [\theta \ 0 \ \dot{\theta} \ 0]$. In PSF, the strain gage is basically ignored and you are left with a servo position control. As illustrated in Figure 3.8, this results in many oscillations in the link deflection every time the servo switches between ± 30 degrees.

□ □ □

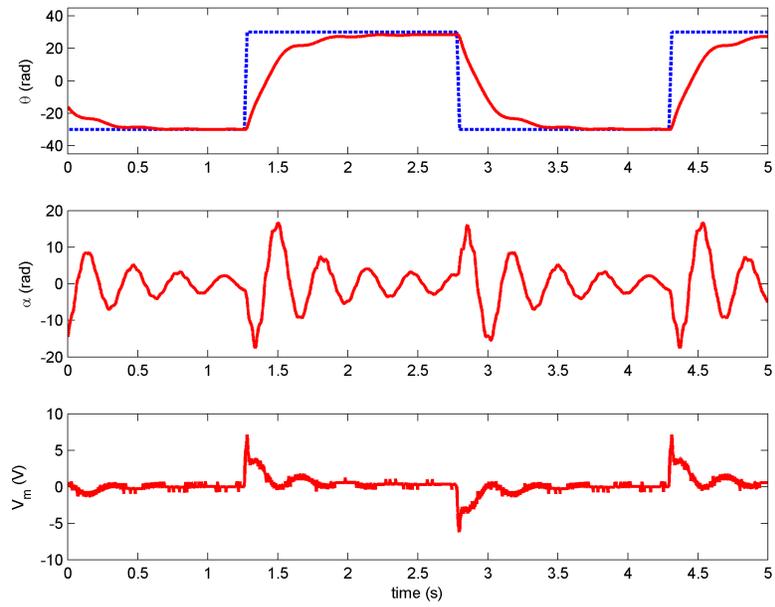


Figure 3.8: Partial-State Closed-Loop Response

3.5 Results

B-6 Fill out Table 3 with your answers from your control lab results - both simulation and implementation.

Description	Symbol	Value	Unit
LQR Simulation			
LQR Weighting Matrices	Q	$\begin{bmatrix} 150 & 0 & 0 & 0 \\ 0 & 1 & 0 & 1 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 3 \end{bmatrix}$	
	R	1	
LQR Gain	K	[12.2 -23.0 1.36 -0.311]	
Settling time	t_s	0.42	s
Percentage overshoot	PO	2.8	%
Maximum deflection	$ \alpha _{\max}$	9.5	deg
LQR Implementation			
LQR Weighting Matrices	Q	$\begin{bmatrix} 125 & 0 & 0 & 0 \\ 0 & 1 & 0 & 1 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 5 \end{bmatrix}$	
	R	1	
LQR Gain	K	[11.2 -30.5 1.46 -0.68]	
Settling time	t_s	0.48	s
Percentage overshoot	PO	1.9	%
Maximum deflection	$ \alpha _{\max}$	9.2	deg

Table 3: Results

4 SYSTEM REQUIREMENTS

Required Software

- Microsoft Visual Studio
- Matlab® with Simulink®, Real-Time Workshop, and the Control System Toolbox.
- QUARC® 2.1, or later.

See the QUARC® software compatibility chart at [5] to see what versions of MS VS and Matlab are compatible with your version of QUARC and for what OS.

Required Hardware

- Data-acquisition (DAQ) card that is compatible with QUARC. This includes Quanser Hardware-in-the-loop (HIL) boards such as:
 - Q2-USB
 - Q8-USB
 - QPID
 - QPIDeand some National Instruments DAQ devices (e.g., NI USB-6251, NI PCIe-6259). For a full listing of compliant DAQ cards, see Reference [1].
- Quanser SRV02-ET rotary servo.
- Quanser Rotary Flexible Joint (attached to SRV02).
- Quanser VoltPAQ-X1 power amplifier, or equivalent.

Before Starting Lab

Before you begin this laboratory make sure:

- QUARC® is installed on your PC, as described in [3].
- The *QUARC Analog Loopback Demo* has been ran successfully.
- SRV02 Rotary Flexible Joint and amplifier are connected to your DAQ board as described Reference [7].

4.1 Overview of Files

File Name	Description
Flexible Link User Manual.pdf	This manual describes the hardware of the Rotary Flexible Link system and explains how to setup and wire the system for the experiments.
Flexible Link Workbook (Student).pdf	This laboratory guide contains pre-lab questions and lab experiments demonstrating how to design and implement a position controller on the Quanser SRV02 Flexible Link plant using QUARC® .
setup.flexgagem	The main Matlab script that sets the SRV02 motor and sensor parameters, the SRV02 configuration-dependent model parameters, and the Flexible Link (i.e., flexgagem) sensor parameters. Run this file only to setup the laboratory.
config_srv02.m	Returns the configuration-based SRV02 model specifications R_m , k_t , k_m , K_g , $\eta_{g_}$, B_{eq} , J_{eq} , and $\eta_{m_}$, the sensor calibration constants K_{POT} , K_{ENC} , and K_{TACH} , and the amplifier limits V_{MAX_AMP} and I_{MAX_AMP} .
config.flexgagem	Returns the Flexible Link model inertial, J_l , viscous damping, B_l , and sensor calibration constant K_{GAGE} .
d_model_param.m	Calculates the SRV02 model parameters K and τ based on the device specifications R_m , k_t , k_m , K_g , $\eta_{g_}$, B_{eq} , J_{eq} , and $\eta_{m_}$.
FLEXGAGE_ABCD.eqns_student.m	Contains the incomplete state-space A, B, C, and D matrices. These are used to represent the Flexible Link system.
calc_conversion_constants.m	Returns various conversions factors.
s.flexgagemdl	Simulink file that simulates the Flexible Link system when using a full or partial state-feedback control.
q.flexgagem_id.mdl	When ran with QUARC® , this Simulink model measures the Flexible Link angle. The measured response can then be used to find the stiffness of the link.
q.flexgagem_val.mdl	This Simulink model is used with QUARC® to compare the Flexible Link state-space model with the measured response from the actual system.
q.flexgagemdl	Simulink file that implements a closed-loop state-feedback controller on the actual FLEXGAGE system using QUARC® .

Table 4: Files supplied with the SRV02 Flexible Link Control Laboratory.

File Name	Description
Flexible Link Workbook (Instructor).pdf	Same as the student version except with solutions.
flexgage.mws	Maple worksheet used to develop the model for the FLEXGAGE experiment. Waterloo Maple 9, or a later release, is required to open, modify, and execute this file.
flexgage.html	HTML presentation of the Maple Worksheet. It allows users to view the content of the Maple file without having Maple 9 installed. No modifications to the equations can be performed when in this format.
d_flexgage_Ks.m	Matlab script outputs a sample measured stiffness value, K_s .
d_flexgage_lqr.m	Matlab script file uses LQR to calculate the control gain K given a the FLEXGAGE model state-space matrices A and B and using a set Q and R matrices.
id_flexgage_Ks.m	Matlab script used to find the stiffness of the flexible link automatically from the free oscillation response found using q_flexgage_id.mdl (or loaded from saved MAT file).
FLEXGAGE_ABCD_eqns.m	Contains the completed state-space A , B , C , and D matrices that represent the Flexible Link system.
plot_flexgage_val.m	Plots the model validation response from q_flexgage_val.mdl (or saved MAT data).
meas_flexgage_specs.m	Plots the step response found in the variables data_theta, data_alpha, and data_vm in the Matlab workspace and measures the corresponding peak time, settling time, percent overshoot, and steady-state error. Users can also use the saved responses contained in MAT files.
meas_step_rsp_specs.m	Function that measures the peak time, settling time, steady-state error, and percent overshoot of a given step response.

Table 5: Instructor design files supplied with the SRV02 Flexible Link Control Laboratory.

File Name	Description
data_flexgage_free_osc.mat	Sample measured free oscillation response of the flexible link.
data_flexgage_val_theta.mat	Sample measured model validation servo response.
data_flexgage_val_theta.mat	Sample measured model validation link response.
data_flexgage_fsf_theta.mat	Sample measured full-state feedback servo response using <i>original LQR</i> controller.
data_flexgage_fsf_alpha.mat	Sample measured full-state feedback link response using <i>original LQR</i> controller.
data_flexgage_fsf_vm.mat	Sample measured full-state feedback input voltage using <i>original LQR</i> controller.
data_flexgage_fsf2_theta.mat	Sample measured full-state feedback servo response using <i>tuned LQR</i> controller.
data_flexgage_fsf2_alpha.mat	Sample measured full-state feedback link response using <i>tuned LQR</i> controller.
data_flexgage_psf_vm.mat	Sample measured full-state feedback input voltage using <i>tuned LQR</i> controller.
data_flexgage_psf_theta.mat	Sample measured partial-state feedback servo response using <i>tuned LQR</i> controller.
data_flexgage_psf_alpha.mat	Sample measured partial-state feedback link response using <i>tuned LQR</i> controller.
data_flexgage_psf_vm.mat	Sample measured partial-state feedback input voltage using <i>tuned LQR</i> controller.

Table 6: Data files supplied with the SRV02 Flexible Link Control Laboratory.

4.2 Setup for Finding Stiffness

Before beginning in-lab procedure outlined in Section 2.3.1, the `q_flexgage_id` Simulink diagram must be properly configured.

Follow these steps:

1. Setup the SRV02 with the Flexible Link module as detailed in the Flexible Link User Manual ([7]).
2. Load the Matlab software.
3. Browse through the *Current Directory* window in Matlab and find the folder that contains the file `q_flexgage_id.mdl`.
4. Open the `q_flexgage_id.mdl` Simulink diagram, shown in Figure 2.4.
5. **Configure DAQ:** Ensure the HIL Initialize block subsystem is configured for the DAQ device that is installed in your system. By default, the block is setup for the Quanser Q8 hardware-in-the-loop board. See Reference [1] for more information on configuring the HIL Initialize block.

4.3 Setup for Model Validation

Before performing the in-lab exercises in Section 2.3.2, the `q_flexgage_val` Simulink diagram and the `setup_flexgage.m` script must be configured.

Follow these steps to get the system ready for this lab:

1. Setup the SRV02 with the Flexible Link module as detailed in [7].
2. Load the Matlab software.

3. Browse through the *Current Directory* window in Matlab and find the folder that contains the QUARC FLEXGAGE file *q_flexgauge_val.mdl*.
4. Open the *q_flexgauge_val.mdl* Simulink diagram, shown in Figure 2.7.
5. **Configure DAQ:** Ensure the HIL Initialize block in the *SRV02 Flexible Link* subsystem is configured for the DAQ device that is installed in your system. By default, the block is setup for the Quanser Q8 hardware-in-the-loop board. See Reference [1] for more information on configuring the HIL Initialize block.
6. **Configure Sensor:** The position of the SRV02 load shaft can be measured using either the potentiometer or the encoder. Set the *Pos Src* Source block in *q_flexgauge_val*, as shown in Figure 2.7, as follows:
 - 1 to use the potentiometer
 - 2 to use the encoder

Note that when using the potentiometer, there will be a discontinuity.

7. **Configure Input:** Set the *Manual Switch* to the DOWN position for a step input or the UP position for a square signal.
8. Open the *setup_flexgauge.m* file. This is the setup script used for the FLEXGAGE Simulink models.
9. **Configure setup script:** When used with the Flexible Link, the SRV02 has no load (i.e., no disc or bar) and has to be in the high-gear configuration. Make sure the script is setup to match this setup:
 - EXT_GEAR_CONFIG to 'HIGH'
 - LOAD_TYPE to 'NONE'
 - Ensure ENCODER_TYPE, TACH_OPTION, K_CABLE, AMP_TYPE, and VMAX_DAC parameters are set according to the SRV02 system that is to be used in the laboratory.
 - CONTROL_TYPE to 'MANUAL'.

For Instructors: Set CONTROL_TYPE = 'AUTO' to automatically set the link stiffness, K_s , load the model, and find the LQR gain.

The students should not have access to the files given in Table 5 and Table 6. However, exactly what should be given to the students is at the discretion of the instructor.

4.4 Setup for Flexible Link Control Simulation

Before going through the control simulation in Section 3.4.1, the *s_flexgauge* Simulink diagram and the *setup_flexgauge.m* script must be configured.

Follow these steps to configure the lab properly:

1. Load the Matlab software.
2. **IMPORTANT:** Make sure the model you found in Section 2.3.2 is entered in FLEXGAGE _ABCD _eqns _student.m.

For Instructors: Set CONTROL_TYPE = 'AUTO' as explained in Section 4.3 to load the model from the FLEXGAGE _ABCD _eqns.m files and then find the LQR gain, K , automatically.
3. Browse through the *Current Directory* window in Matlab and find the folder that contains the *s_flexgauge.mdl* file.
4. Open *s_flexgauge.mdl* Simulink diagram shown in Figure 3.2.
5. Configure the *setup_flexgauge.m* script according to your hardware. See Section 4.3 for more information.
6. Run the *setup_flexgauge.m* script.
7. Enter the stiffness (K_s) you found in Section 2.3.1.

4.5 Setup for Flexible Link Control Implementation

Before beginning the in-lab exercises given in Section 3.4.2 (or Section 3.4.3), the `q_flexgag` Simulink diagram and the `setup_flexgag.m` script must be setup.

Follow these steps to get the system ready for this lab:

1. Setup the SRV02 with the Flexible Link module as detailed in [7] .
2. Load the Matlab software.
3. Browse through the *Current Directory* window in Matlab and find the folder that contains the `q_flexgag.mdl` file.
4. Open the `q_flexgag.mdl` Simulink diagram shown in Figure 3.5.
5. **Configure DAQ:** Ensure the HIL Initialize block in the *SRV02 Flexible Link* subsystem is configured for the DAQ device that is installed in your system. By default, the block is setup for the Quanser Q8 hardware-in-the-loop board. See Reference [1] for more information on configuring the HIL Initialize block.
6. **Configure Sensor:** The position of the SRV02 load shaft can be measured using the potentiometer or the encoder. Set the *Pos Src* Source block in `q_flexgag`, as shown in Figure 3.5, as follows:
 - 1 to use the potentiometer
 - 2 to use the encoder

Note that when using the potentiometer, there will be a discontinuity.

7. **Configure setup script:** Set the parameters in the `setup_flexgag.m` script according to your system setup. See Section 4.3 for more details.
8. Run the `setup_flexgag.m` script.

5 LAB REPORT

This laboratory contains two groups of experiments, namely,

1. Modeling the Quanser Rotary Flexible Link system, and
2. State-feedback control using LQR.

For each experiment, follow the outline corresponding to that experiment to build the *content* of your report. Also, in Section 5.3 you can find some basic tips for the *format* of your report.

5.1 Template for Content (Modeling)

I. PROCEDURE

1. *Finding Stiffness*

- Briefly describe the main goal of the experiment.
- Briefly describe the experiment procedure (Section 2.3.1)

2. *Model Validation*

- Briefly describe the main goal of the experiment.
- Briefly describe the experiment procedure (Section 2.3.2)

II. RESULTS

Do not interpret or analyze the data in this section. Just provide the results.

1. Free-oscillation plot from step 3 in Section 2.3.1.
2. Model validation plot from step 13 in Section 2.3.2.
3. Provide applicable data collected in this laboratory (from Table 2).

III. ANALYSIS

Provide details of your calculations (methods used) for analysis for each of the following:

1. Measured link stiffness in step 5 in Section 2.3.1.
2. Model discrepancies given in step 15 in Section 2.3.2.

IV. CONCLUSIONS

Interpret your results to arrive at logical conclusions for the following:

1. How does the model compare with the actual system in step 14 of Section 2.3.2, *State-space model validation*.

5.2 Template for Content (Control)

I. PROCEDURE

1. *Simulation*

- Briefly describe the main goal of the simulation.
- Briefly describe the simulation procedure (Section 3.4.1).
- Briefly describe the procedure in step 4 of Section 3.4.1 to examine the effect of variables on the gain and closed-loop response.
- Briefly explain the procedure used to find Q and R in step 8 of Section 3.4.1).

2. *Full-State Feedback Implementation*

- Briefly describe the main goal of this experiment.
- Briefly describe the experimental procedure (Section 3.4.2).
- Briefly explain the procedure used to tune the controller in step 12 of Section 3.4.3.

3. *Partial-State Feedback Implementation*

- Briefly describe the main goal of this experiment.
- Briefly describe the experimental procedure (Section 3.4.3).

II. RESULTS

Do not interpret or analyze the data in this section. Just provide the results.

1. Response plot from step 6 in Section 3.4.1, *Full-state feedback LQR controller simulation*.
2. Response plot from step 7 in Section 3.4.2, for *Full-state feedback LQR controller implementation*.
3. Response plot from step 10 in Section 3.4.2, for *Tuned LQR full-state feedback controller implementation*.
4. Response plot from step 6 in Section 3.4.3, for *Partial-state feedback LQR controller implementation*.
5. Provide applicable data collected in this laboratory (from Table 3).

III. ANALYSIS

Provide details of your calculations (methods used) for analysis for each of the following:

1. Settling time and percent overshoot in step 7 in Section 3.4.1, *Full-state feedback LQR controller simulation*.
2. Settling time and percent overshoot in step 8 in Section 3.4.2, for *Full-state feedback LQR controller implementation*.
3. Settling time and percent overshoot in step 11 in Section 3.4.2, for *Tuned LQR full-state feedback controller implementation*.
4. Comparison between partial-state and full-state feedback in step 7 in Section 3.4.3.

IV. CONCLUSIONS

Interpret your results to arrive at logical conclusions for the following:

1. Whether the controller meets the specifications in step 7 in Section 3.4.1, *Full-state feedback LQR controller simulation*.
2. Whether the controller meets the specifications in step 8 in Section 3.4.2, for *Full-state feedback LQR controller implementation*.
3. Whether the controller meets the specifications in step 11 in Section 3.4.2, for *Tuned LQR full-state feedback controller implementation*.

5.3 Tips for Report Format

PROFESSIONAL APPEARANCE

- Has cover page with all necessary details (title, course, student name(s), etc.)
- Each of the required sections is completed (Procedure, Results, Analysis and Conclusions).
- Typed.
- All grammar/spelling correct.
- Report layout is neat.
- Does not exceed specified maximum page limit, if any.
- Pages are numbered.
- Equations are consecutively numbered.
- Figures are numbered, axes have labels, each figure has a descriptive caption.
- Tables are numbered, they include labels, each table has a descriptive caption.
- Data are presented in a useful format (graphs, numerical, table, charts, diagrams).
- No hand drawn sketches/diagrams.
- References are cited using correct format.

6 SCORING SHEET FOR PRE-LAB MODELING QUESTIONS

Student Name :

Question ¹	A-1	A-2	A-3
1			
2			
3			
4			
5			
6			
7			
8			
Total			

¹This scoring sheet is for the Modeling Pre-Lab questions in Section 2.2.

7 SCORING SHEET FOR LAB REPORT (MODELING)

Student Name:

Item ¹	CONTENT						FORMAT	
	K-2	B-5	B-6	B-7	B-8	B-9	GS-1	GS-2
I. PROCEDURE								
I.1. Finding Stiffness								
1								
I.2. Model Validation								
2								
II. RESULTS								
1								
2								
3								
III. ANALYSIS								
1								
2								
IV. CONCLUSIONS								
1								
Total								

¹This scoring sheet corresponds to the report template in Section 5.1.

8 SCORING SHEET FOR PRE-LAB CONTROL QUESTIONS

Student Name :

Question ¹	A-1	A-2	A-3
1			
2			
3			
4			
Total			

¹This scoring sheet is for the Control Pre-Lab questions in Section 3.3

9 SCORING SHEET FOR LAB REPORT (CONTROL)

Student Name:

Item ¹	CONTENT									FORMAT	
	K-1	K-3	B-2	B-4	B-5	B-6	B-7	B-9	GS-1	GS-2	
I. PROCEDURE											
I.1. Simulation											
1											
2											
3											
I.2. Full-State Feedback Implementation											
1											
2											
I.3. Partial-State Feedback Implementation											
1											
II. RESULTS											
1											
2											
3											
4											
5											
III. ANALYSIS											
1											
2											
3											
4											
IV. CONCLUSIONS											
1											
2											
3											
Total											

¹This scoring sheet corresponds to the report template in Section 5.2.

APPENDIX A

INSTRUCTOR'S GUIDE

Every laboratory in this manual is organized into four sections.

Background section provides all the necessary theoretical background for the experiments. Students should read this section first to prepare for the Pre-Lab questions and for the actual lab experiments.

Pre-Lab Questions section is not meant to be a comprehensive list of questions to examine understanding of the entire background material. Rather, it provides targeted questions for preliminary calculations that need to be done prior to the lab experiments.

Lab Experiments section provides step-by-step instructions to conduct the lab experiments and to record the collected data.

System Requirements section describes all the details of how to configure the hardware and software to conduct the experiments. It is assumed that the hardware and software configuration have been completed by the instructor or the teaching assistant *prior* to the lab sessions. However, if the instructor chooses to, the students can also configure the systems by following the instructions given in this section.

Assessment of ABET outcomes is incorporated into this manual as shown by indicators such as **A-1, A-2**. These indicators correspond to specific performance criteria for an outcome. The SRV02 Instructor Lab Manual (Reference [6] for Appendix A) provides extensive explanations on how to incorporate outcomes assessment into your course and how to use the indicators built into the curriculum.

REFERENCES

- [1] Quanser Inc. *QUARC User Manual*.
- [2] Quanser Inc. *SRV02 QUARC Integration*, 2008.
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- [5] Quanser Inc. *QUARC Compatibility Table*, 2010.
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- [7] Quanser Inc. *SRV02 Rotary Flexible Link User Manual*, 2011.
- [8] Norman S. Nise. *Control Systems Engineering*. John Wiley & Sons, Inc., 2008.