



On the Physical and Combustion Properties of Hydrogen and the Feasibility and Characteristics of Hydrogen Fuel-Cell Vessels

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Sandia HQ: <mark>Albuquer</mark>que NM



Hydrogen and Fuel Cells Program

- Sandia is the largest National Lab in the U.S.
 - U.S. Department of Energy (DOE) ~13,000 employees
 - ~ US \$3.2B/yr from DOE, other federal agencies, and private industry
 - H₂ Program in Livermore, CA (SF Bay Area)
- Hydrogen program: 60+ years of work, in a wide range of areas (H₂ storage, production, delivery, development of regulations, market transformation), which we apply to enable impactful clean energy solutions
- Market Transformation: Zero Emission H₂/Fuel Cell Maritime Program:





Livermore CA (SF Bay Area)



H₂ molecule

Natural Gas

(90% CH₄)

Water

 (H_2O)



Hydrogen Properties:

- Is typically a gas, but can be a liquid (LH₂) if made very cold (20 K).
- LH₂ evaporates very fast (4,000 gallons will evaporate in ~7 seconds)
- More buoyant than helium. Goes straight up at ~40 mph.

Overall, H_2 is very similar to natural gas (which is ~ 90% methane, CH_4).

H₂ is NOT a Greenhouse Gas, unlike natural gas which is a potent GHG.

If spilled, LH₂ evaporates from the water leaving no residue.

methane reforming

> water electrolysis

 H_2 can be ignited given an ignition source and the right H_2 /air mixture.

Energetically, a kg of H_2 has about the same energy as a gallon of diesel fuel.





When hydrogen is used in a *Fuel Cell* it produces ZERO pollution or greenhouse gas at point of use





$2 \text{ H}_2 + \text{O}_2 \rightarrow 2 \text{ H}_2\text{O}$

- -- commercially available
- -- more energy efficient than diesel generators
- -- eliminates emissions at the point of use
- -- eliminates fuel spills, greatly reduces noise
- -- emissions can only arise from H₂ production/delivery
- -- no "thermal runaway" possible

<u>Going In:</u> H₂ and air

<u>Going Out:</u> Electricity Waste Heat Warm humidified air



 LH_2 :

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Physical Properties of LH₂ and LNG

Note: 75 °F = 297 K

Liquid Normal Boiling Point = 20 K (-253 °C). Liquid Density = 71 g/L Lower Heating Value = 120 MJ/kg

LNG (LCH₄):

Liquid Normal Boiling Point = 111 K (-162 °C). Liquid Density = 422 g/L Lower Heating Value = 45 MJ/kg

LH₂ and LNG are stored in similar ways:



LH₂ Storage Tank on Trailer



LNG Storage Tank on Trailer

LH₂ has 0.38 times the mass of LNG, but has 2.4 times the volume, per unit of energy (LHV).

LH₂ has 0.36 times the mass of diesel, but has 4.2 times the volume, per unit of energy (LHV).







LH₂ Evaporates Faster than LNG, Cools Surfaces Less (Spill)



LNG (CH_4):



H₂ molecules barely interact at all

Heat of Vaporization = 0.922 kJ/mole

Heat of Vaporization = 8.5 kJ/mole

-- For equal amounts of stored energy, LH_2 takes <u>3 times less</u> energy than LNG to evaporate. In a spill, LH_2 will cool surroundings much less than LNG.

-- We estimate that if we spilled 1200 kg of LH_2 , it would take ~ 6 seconds to evaporate



Let's Pause for Some Combustion Definitions

Explosion or Detonation: Extremely fast combustion where the flame propagates through the unburned fuel/air mix at supersonic speeds (~ 1000 m/s). Explosions produce loud bangs and very damaging overpressures. **Explosions are very, very bad**.

Deflagration: Fast combustion where the flame propagates through the unburned fuel/air mix at subsonic speeds (~ 100 m/s). Deflagrations are not as loud, and have 10x less damaging overpressures than explosions. **Deflagrations are very bad.**

Fire: Ordinary combustion where the flame propagates through the unburned fuel/air mix at slow speeds (~ 10 m/s). Fires are not loud, have negligible overpressure. **Fires are bad.** If we work to prevent fires, we also prevent explosions and deflagrations.

Weak (Thermal) Ignition Sources: Matches, sparks, hot surfaces, open flames (< 50 mJ). These are typically the ignition sources that cause accidents.

Strong (Shock Wave) Ignition Sources : blasting caps, TNT, high-voltage capacitor shorts (exploding wires), lightning (> 4 MJ). Strong sources are $\sim 10^8$ stronger than weak initiators.





Gaseous Flammability of H₂ vs NG

Both H_2 and NG (methane) mixtures with air are easily ignited by "Weak (Heat) Ignition Sources" such as: sparks, hot surfaces, open flames (< 50 mJ).

Fire regulations reference the "Lower Flammability Limit" (LFL) because the greatest fire risk comes from building up gas in initially clean air:

Definition: % by volume = [Volume (Fuel)/Volume (Fuel + Air)] x 100.

The LFL to upper flammability limit (UFL) for $H_2 = 4.0 - 75.0$ % at RT. The LFL to UFL of methane is = 5.3 - 15.0 % at room temp. Note: the LFL for diesel fuel is 0.6%.

The minimum ignition energy for $CH_4 = 0.29 \text{ mJ}$; That for H_2 is 0.02 mJ. Static discharges from human beings are ~ 10 mJ, so both CH_4 and H_2 ignite easily.

*H*₂ and NG are both easily ignited by weak ignition sources, and start to burn at similar lower flammability limits



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Direct Explosions/Detonations of H₂ vs NG

The lower explosion limit (LEL) of H_2 at room temperature (% by volume) - upper explosion limit (UEL) = 18.3 – 59.0 % at RT. The LEL to UEL of methane is = 6.3 – 13.5 % at room temperature.

Experiments show that both H₂ and CH₄ require three things to directly detonate:

- **1. The fuel/air mix in the LEL UEL range.**
- 2. A strong ignition source
- 3. Confinement . Unconfined vapors of H₂ and CH₄ WILL NOT directly detonate, explode or deflagrate.

A.D. Little study of LH_2 for the U.S. Air Force: For a series of LH_2 spills up to 5,000 gallon (1342 kg), igniting the unconfined vapor above the pool with spark, flame, or explosive charges produced only fire.

-- A.D. Little, Inc., Report to the U.S. Air Force, C-61092, (1960).

-- L.H. Cassutt, Report to the U.S. Air Force, Report No. 61-05-5182, (1964).

Sandia "Phoenix" Tests of LNG: Ignition of unconfined vapors above a 30,000 gallon pool of LNG produced only fire.



Deflagration to Detonation Transition (DDT)

In a confined environment with a lot of obstacles or internal structures, it is possible to get an explosion or detonation even with weak ignition sources.

Unlike direct detonation, which requires a strong ignition source, this type of explosion can start with a normal fire. In the confined/obstructed environment, the speed of the combustion accelerates over time and distance to a deflagration. With further acceleration, the deflagration transitions to a detonation because of turbulent mixing of the confined gases caused by the obstructions. We have a Deflagration to Detonation Transition (DDT).

For H_2 , DDT can only occur for 12% fuel /air mix or higher. Both H_2 and NG can experience DDT, although it is easier for hydrogen. For hydrogen, ~ 10 m of run-up distance is required for the DDT.





Hindenburg Accident (1937)





It was a fire, NOT an explosion. Explosions are fast (think firecracker). The airship initially stayed aloft while burning. But fires are bad too and need to be prevented!

The method of storing hydrogen for the airship (rubberized gas bags) bears no resemblance to the highly engineered DOT-approved stainless steel LH₂ tanks in use today.



Improved to:



NASA mastered hydrogen, the "signature fuel" of the American Space Program since the 1960s.





The Nature of Fires: LH₂ vs. LNG

NASA funded a model study of the fire safety aspects of LH_2 and LNG as part of their program in alternative-fuel aircraft in the 1980s.



 $6 \text{ m}^3 = 6,000 \text{ L} = 1587$ gallons

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= 426 kg of LH_2
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= 2532 kg of LNG

A.D. Little, Inc., "An assessment of the crash fire hazard of liquid hydrogen fueled aircraft." Final Report to the National Aeronautics and Space Administration,NASACR-165526. 1982.

 LH_2 pool fires burn out faster than LNG pool fires due to the much lower heat of vaporization of LH_2 . But both fuels, if spilled, burn themselves out rapidly.



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Burning of H₂:

LH₂ Fires Pose Less Safety Risk Than LNG Fires

Results from the NASA-funded study:



hazard distance (5kW/m²) as a function of the heat content of the released fuel.

LH₂ fires are radiatively much less hazardous than LNG fires for two reasons:
1. They don't contain carbon, and so radiate less than NG flames.
2. Since the product of H₂ combustion is H₂O, the flame radiation is strongly absorbed by H₂O in the air.





Hydrogen Embrittlement





H can accumulate at crack points, promoting crack propagation and eventual material failure in certain high-strength alloys and steels.

<u>Solved Problem</u>: The industry practice is to use 304 or 316 stainless steel in all H_2 plumbing, which is satisfactory.

But it is very important to ensure these alloys are used!





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HYDROGEN ENERGY

Comparison of the safety-related physical and combustion properties of liquid hydrogen and liquid natural gas in the context of the SF-BREEZE high-speed fuel-cell ferry

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We review liquid hydrogen (1H2) as a maritime vessel fuel, from descriptions of its fundamental properties to its practical application and safety aspects, in the context of the San Francisco Bay Renewable Energy Electric Vessel with Zero Emissions (SF-BREEZE) highspeed fuel-cell ferry. Since marine regulations have been formulated to cover liquid natural gas (LNG) as a primary propulsion fuel, we frame our examination of LH2 as a comparison to LNG, for both maritime use in general, and the SF-BREEZE in particular. Due to weaker attractions between molecules, LH2 is colder than LNG, and evaporates more easily. We describe the consequences of these physical differences for the size and duration of spills of the two cryogenic fuels. The classical flammability ranges are reviewed, with a focus on how fuel buoyancy modifies these combustion limits. We examine the conditions for direct fuel explosion (detonation) and contrast them with initiation of normal (laminar) combustion. Direct fuel detonation is not a credible accident scenario for the SF-BREEZE. For both fuels, we review experiments and theory elucidating the deflagration to detonation transition (DDT). LH2 fires have a shorter duration than energy-equivalent LNG fires, and produce significantly less thermal radiation. The thermal (infrared) radiation from hydrogen fires is also strongly absorbed by humidity in the air. Hydrogen permeability is not a leak issue for practical hydrogen plumbing. We describe the chemistry of hydrogen and methane at iron surfaces, clarifying their impact on steel-based hydrogen storage and transport materials. These physical, chemical and combustion properties are pulled together in a comparison of how a LH2 or LNG pool fire on the Top Deck of the SF-BREEZE might influence the structural integrity of the aluminum deck. Neither pool fire scenario leads to net heating of the aluminum decking. Overall, LH2 and LNG are very similar in their physical and combustion properties, thereby posing similar safety risks. For ships utilizing LH2 or LNG, precautions are needed to avoid fuel leaks, minimize ignition sources, minimize confined spaces, provide ample ventilation for required confined spaces, and to monitor the enclosed spaces to ensure any fuel accumulation is detected far below the fuel/air mix threshold for any type of combustion.

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More information about hydrogen gas (H_2) and liquid hydrogen (LH_2) can be found in:

L.E. Klebanoff et al., International Journal of Hydrogen Energy **42**, 757 (2017).

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Hydrogen Technology is Here and Growing







Fuel Cell Electric Vehicles Can:

- ✓ Refuel in 5 minutes
- ✓ Have a 366 mile driving range (limited by H_2 storage)
- $\checkmark\,$ Only clean water vapor as the tailpipe emissions
- $\checkmark\,$ No need to plug in.

And it's not just fuel cell cars:



Fuel Cell Forklifts

TOACH HE CE CERCENT CONTRACTOR

Fuel Cell Buses

Fuel Cell Lighting

H₂ Stations are Being Built

FORNIA

San Dieg





H₂ Has Been Delivered and Used for Decades



Today: AC Transit Bus Station, Emeryville CA



1964 - 1973



1981 - 2011

A typical LH₂ trailer can deliver 4000 kg (~15,000 gallons) at a time. (1 kg LH₂ = 3.72 gallons)

Trailer LH₂ tanks are DOT-approved and have never been breached in a road accident.







Densities of Hydrogen:

700 bar gas = 40 g/L

 LH_2 (at 1 bar pressure) = 71 g/L

Solid $AIH_3 = 149 \text{ g/L}$

The low density of even high pressure H_2 limits the range of fuel cell vehicles. Solid-state H_2 storage could be better!

But solid-state hydrogen storage is still a R&D area.





SF-BREEZE: The first study to show that H₂ fuel cells can be used in maritime propulsion, and how to do it.





Route: San Francisco to Vallejo, CA

	Ferry	Hydrogen Station
Technical	\checkmark	\checkmark
Regulatory	\checkmark	\checkmark
Economic	Higher than conventional now, today's market acceptance to be determined	





Project Integrates Ship Designers, Regulators, H₂ Experts and End Users







USCG MSC and Design and Eng. Stds.



USCG Sector San Francisco



Sandia

National

Laboratories

USCG Liquid Gas Carrier NCOE





American Bureau of Shipping



Work Funded by The U.S. Department of Transportation (DOT), Maritime Administration (MARAD) through MARAD's Maritime Environmental and Technical Assistance (META) program.





SF-Breeze Design







Design generated by Elliott Bay Design Group

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SF-BREEZE details



- Fuel: ~2,000 kg LH₂ per day
- Propulsion power 4.4 MW, installed: 4.92 MW
- Passengers: 150
- Service Speed: 35 knots
- Length 109' x Beam 33' x Depth 11.25'
 Full Load Draft ~ 4.6'
- Emissions: Zero
- Fuel Spills: Zero







The SF-BREEZE Project Led to the Zero-V Hydrogen Fuel Cell Research Vessel

Overall Feasibility Question: Is it technically and economically possible to create a zero-emissions H_2 fuel cell research vessel that meets or <u>exceeds</u> the requirements of such vessels operating along U.S. coastlines?



Gerd Petra Haugom (L) Hans-Christian Wintervoll DNV GL



Lennie Klebanoff Sandia National Laboratories



Bruce Appelgate Scripps Institution of Oceanography



Zoltan Kelety Scripps Institution of Oceanography



Glosten Participants: (L-R) Ian McCauley, Sean Caughlan, Robin Madsen and Catherine Farish.





Scripps Missions Define the Zero-V Performance



The Zero-V has very different performance needs:

- Desired calm water speed: 10 knots (instead of 35 knots for the SF-BREEZE)
- Desired range: 2,400 nautical miles (instead of 100 nm for the SF-BREEZE)
- Endurance: 14 days (instead of 4 hours for the SF-BREEZE).

A zero-emission research vessel is feasible NOW using existing technology





 Oceanographic research vessel for coastal / regional operations

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DNVG

- Uses clean hydrogen: No fossil fuels!
- Zero emissions: Clean/no GHGs!
- Carries no diesel: No oil spills!
- All-electric propulsion: Quiet!
- FEASIBLE with existing technology
- Outstanding scientific capabilities
- Advanced instrumentation
- Designed for California's educational and R&D needs

A bold, transformative gamechanger

The zero-emission research vessel (Zero-V) concept vessel has a range of 2,400 nm, speed of 10 knots, with berths for up to 20 scientists, supporting generalpurpose missions. Anticipated cost to build: \$80 million.





Hazardous Zone Plan for the Zero-V Drives The Design







The GHG Reduction from Using H₂ technology REALLY depends on How the H₂ is Made



-- the equivalent GHG emissions for diesel fuel is 87.4 grams CO₂ (eq.)/MJ_{fuel}

More information on the calculation of GHG emissions from H_2 fuel cell technology can be found it: L.E. Klebanoff , J.W. Pratt et al., Transportation Research D 54, 250 (2017).



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Emissions from LH₂ Production (the Zero-V itself is zero-emissions):



Using H_2 from any source, dramatic reductions in criteria pollutants below Tier 4 are provided. Using renewable hydrogen, a 91% reduction in CO₂ (eq.) emissions is obtained.



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SANDIA REPORT SAND2018-4664 Unlimited Release | Printed May 2018

Feasibility of the Zero-V:

A Zero-Emission, Hydrogen Fuel-Cell, Coastal Research Vessel

Leonard E. Klebanoff, Joseph W. Pratt, Robert T. Madsen, Sean A.M. Caughlan, Timothy S. Leach, T. Bruce Appelgate, Jr., Stephen Zoltan Kelety, Hans-Christian Wintervoll, Gerd Petra Haugom and Anthony TX. Teo

Prepared by Sandia National Laboratories, Livermore, California 94550

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Dr. Bruce Appelgate (Scripps) is currently seeking funding for further design and construction of the Zero-V.

maritime.sandia.gov

Work Funded by The U.S. Department of Transportation (DOT), Maritime Administration (MARAD) through MARAD's Maritime Environmental and Technical Assistance (META) program.





A H₂ Fuel Cell Ferry Is Under Construction!

-- Funded by the State of California Air Resources Board (CARB)



The world's first commercial hydrogen fuel cell ferry, and first hydrogen fuel cell vessel in the U.S.

- Aluminum catamaran
- 70' long
- 84 passenger (reconfigurable)
- 22 knot top speed

Project Lead



Funding & Administration





AIR OUALITY



This project is supported by the "California Climate Investments" (CCI) program



H₂ Vessel Feasibility Questions Encountered and Passed

- Will they float? 🗸
- Can they go fast enough, up to 35 knots?
- Can they carry a decent number of people (~150)?
- Do they have sufficient range before needing refueling?
- Can the hydrogen suppliers provide 2500 kg of LH₂ per day?
- Can the hydrogen suppliers provide renewable LH_2 ?
- Can they be refueled fast enough for commuter service ?
- Would the technology be supported by Bay Area Ports?
- Are there deep cuts in well-to-waves (WTW) GHG emissions?
- Are there deep cuts in WTW criteria pollutant emissions? \checkmark
- Can they satisfy regulatory requirements to gain an Approval in Principal?
- Would the U.S. Coast Guard find any "show stopping" issues?
- Would it be commercially attractive? **TBD**
- Can suitable refueling sites be found for these vessels?
- Would there be support from local government (City Hall, others)?

Hydrogen and Fuel Cells Prog







Hydrogen Storage Technology Materials and Applications



Edited by Lennie Klebanoff --published by CRC Press in 2012

Topics:

- Why we need H₂-based energy
- H₂ Energy Conversion Devices
- All methods of H₂ Storage
- Engineered H₂ Storage Systems
- H₂ Codes and Standards





















Thanks to all my friends and colleagues!







Sujit Ghosh, MARAD

An extra special <u>Thank You</u> to Sujit Ghosh and the US DOT / Maritime Administration (MARAD) for supporting these studies





For more information on H₂ Storage Projects, visit: https://hymarc.org/

For more information on H₂/Fuel Cell Maritime Projects visit: https://maritime.sandia.gov

- Past and current maritime projects
- Download reports

Thank You!

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Extra Slides





Solid-State H₂ Storage Materials



 MH_x : FeTi -H, LaNi₅-H, NaAlH₄

Benefits:

- Can reduce tank pressures from 10,000 psi to 100 psi (safer).
- ✓ Can triple the storage density of 700 bar H_2 .

But, something is wrong with all of the MHx, making this an R&D area:

- 1. Too heavy
- 2. Don't release hydrogen fast enough (bad kinetics)
- 3. Don't re-absorb hydrogen fast enough (bad kinetics)

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- 4. Bind hydrogen too tightly (bad thermodynamics)
- 5. The H₂ they release is contaminated with other species
- 6. Too expensive

MARC (

7. Degrade when cycled



-- funded by DOE EERE FCTO to investigate the fundamental limitations of hydrogen storage materials.





H₂ Aircraft Technology

-AH ^o _{Combustion} (kJ/g) (LHV)		
120.0		
50.1		
42.8		



Prior Studies:

- Lockheed studies of the 1970's
- > 2002 Airbus Study (Cryoplane)
- Boeing Study: (2006)
 - -- Published in Proceedings of the 25TH INTERNATIONAL CONGRESS OF THE AERONAUTICAL SCIENCES (2006)





Larger Fuselage Accommodates LH₂ Fuel, Cargo, People





- From "Hydrogen Aircraft Technology", pg. 46





Comparison of Jet-A vs. LH₂-Fueled Subsonic Passenger Aircraft (Mach 0.85, 400 passengers, Range = 5,500 nmi)

			Ratio
	Jet-A	LH ₂	(LH ₂ /Jet-A)
Takeoff Gross Weight (lb)	523,200	391,700	0.75
Operating Empty Weight (lb)	244,420	242,100	0.99
Fuel Weight (lb)	190,800	61,600	0.32
Wing Area (ft ²)	4,186	3,363	0.80
Take-off Distance (ft)	7,990	6,240	0.78
Energy Utilization $\left(\frac{Btu}{seatnmi}\right)$	1,384	1,239	0.89

Tables 3-11 and 3-12 from "Hydrogen Aircraft Technology"



Potential Military Uses of LH₂ Aircraft

Fighters: No, they only carry ~ 10,000 – 25,000 lbs of fuel, -- performance penalties associated with larger fuselage.

Large Transports: YES

Ex: Lockheed Martin C-5 A/B Galaxy (M0.79, Range 2400 nmi with 263,000 lb payload) C-5 Fuel Capacity = 332,500 lbs.

Note: Transport aircraft consume 80% of fuel used by the U.S. Air Force



High Altitude Long Endurance (HALE) Aircraft: YES

Satcom restoration, communications relay, persistent ISR (Intelligence, Surveillance, Reconnaissance)

- Capabilities: 4 days, 14,000 miles, 450 lb payload, LH₂

Boeing Phantom Eye





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F22

